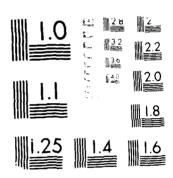
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Software Supportability Risk Assessment in OT&E: Historical Baselines for Risk Profiles

Volume II



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October 7, 1985

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Historical Baselines for Risk Profiles (Volumes I and II)

Item 19 (cont'd):

systems. The data collected was used to develop historical profiles of the activities observed. These profiles are the risk baselines against which negative outcomes can be determined from evaluations of software support risk.

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APPENDIX C

SITE

SURVEY

FORM

04/29/85

DRAFT 7

Site Survey Information

Survey data is needed to determine a historical basis of softwarmaintenance activity across several sites which support Air Force systems.

Briefly, the survey data to be collected includes:

- a) Background data on each software system;
- b) A high-level, subjective assessment by site personnel of the adequacy (product, environment, life cycle process) of the support for each major software system;
- c) Actual software maintenance data (corrections, enhancements, conversions) for each software system in as much detail as is available. Information on each software problem corrected in each block release since "delivery" is needed. This information will be collected during the site visit.

This data will be used to determine:

- a) the availability and consistency of such data;
- b) the effort required to collect such data;
- c) the utility of the data for use in a proposed software supportability risk assessment methodology;
- d) the potential for derivation of a general data collection format for software maintenance data based upon the availability, consistency, effort, and utility as above.

Software systems for which data is desired are indicated in an accompanying list. Other suggested systems for which data might be available will be added to the list as time to collect such data permits. Typically, it should require no more than 30 minutes of a senior software person's time to complete the information for each software system. During the on-site visit it would be beneficial to talk with each of the senior personnel completing the survey form, as well as the appropriate personnel maintaining the configuration management status accounting information. In this way, problems with the assessment data can be resolved and maintenance data which is available on each system can be efficiently collected.

DEFINITIONS FOR SITE SURVEY

- ATE Automatic Test Equipment
- CSCI Computer Software Configuration Item
- ICC Initial Operational Capability
- MA Maintenance Action
- OFP Operational Flight Program
- PMRT Program Management Responsibility Transfer
- S/W Software
- Software System A set of software (specifications, programs, and data) which constitutes a well-defined major function or group of functions. Typical systems include avionics OFP, ground based communications, missile guidance, simulation, threat generator, ATE, and electronic warfare.
- Software Delivery That point in the software life cycle when the software support function assumes responsibility for the "next" set of configuration changes to the software (e.g., next block release). This point is logically no later than PMRT, but could be as early as IOC. This applies when a contractor or government agency assumes the software support function.
- Software Life Cycle Process Management The policy, methodology, procedures, and guidelines applied in a software environment to the software development and support life cycle activities.
- Software Configuration Management A discipline applying technical and administrative direction and surveillance to (1) identify and document the functional and physical characteristics of a configuration item, (2) control changes to those characteristics, and (3) record and report change processing and implementation status.
- Software Maintenance Project Management The software life cycle process management applied during the support phase for the software to accomplish specific software maintenance tasks which derive from software problem reports or change requests.
- Software Maintainability The ease with which software can be changed in order to: correct errors, add or modify system capabilities through software changes, delete features from programs, and modify software to be compatible with hardware changes.

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- Software Supportability A measure of the adequacy of personnel, resources and procedures to facilitate: modifying and installing software, establishing an operational software baseline, meeting user requirements.
- Support Personnel A general term for personnel (military, DoD civilian, or DoD contractor) whose skills are necessary to directly support mission critical system software maintenance. Includes but is not limited to management, technical, non-technical support, and contractor personnel.
- Support System The automated system used to change, test, or manage the configuration of mission critical system software and associated documentation. Includes but is not limited to Host Processor, Software Bench, Laboratory-Integrated Test Facility, Operation-Interated Test Facility, and Configuration Management System.
- Support Facility The physical facility resources that must be available for the software support resources to accomplish a specific task(s).
- Documentation All of the written work describing operating and maintenance procedures for a system.
- Source Code The form of the program code in its source language.
- Consistency A measure of the extent the software products correlate and contain uniform notation, terminology, and symbology.
 - escriptiveness A measure of the extent that software products contain information regarding its objectives, assumptions, inputs, processing, outputs, components, revision status, etc.
- Expandability A measure of the extent that a physical change to information, computational functions, data storage, or execution time can be easily accomplished once the nature of what is to be changed is understood.
- Instrumentation A measure of the extent that software products contain sids which enhance testing.
- Modularity A measure of the extent that a logical partitioning of software products into parts, components, and/or modules has occurred.
- Simplicity A measure of the extent that software products reflect the use of singularity concepts and fundamental structures in organization, language, and implementation techniques.
- Time to Complete MA The time from formal notification (e.g., receipt of anomaly report or software change request) of a software maintenance request to the final disposition of that request (e.g., change is integrated into the next release, or request is denied).

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- Raseline Software Supportability Profile The set of 27 pairs of numbers (or any subset) determined by specifying the (time to complete request, number of requests per unit time) pair for each request category. A request category is the triple (type, priority, complexity) where type is conversion, enhancement, or correction; priority is emergency, urgent, or normal; and complexity is high, medium, low.
- Emergency MA an MA requiring all available personnel's dedicated effort to correct the problem as soon as possible (e.g., 24 hours);
 MIL-STD-1679 severity code 1 or 2: mission termination or severe degradation
- Urgent MA an MA requiring next "block release" turnaround; MIL-STD-1679 severity code 3: mission impact
- Normal MA an MA not in the Emergency or Urgent categories; MIL-STD-1679 severity code 4 or 5: mission inconvenience
- High Complexity MA an MA where changes are in requirements, design, code, and test; or > 10% of CSCI is affected; or several modules are affected by the change (global changes); or the technical nature of the change requires highly specialized personnel skills; or the level of effort by personnel is large
- Medium Complexity MA an MA where changes are in design, code and test; or > 1% of CSCI is affected; or at least two modules are affected by the change (semi-local); or the level of effort by personnel is average
- Low Complexity MA an MA where changes are isolated to only one unit (e.g., one module/compilation unit) of code; or no more than 1% of CSCI is affected; or the level of effort by personnel is minimal
- Conversion (Adaptive) MA Any change/effort to a software system which is initiated as a result of changes in the environment(e.g., hardware, system software) in which the software system must operate.
- Enhancement (Perfective) MA Any change, insertion, deletion, modification, extension, and enhancement made to a software system to meet the evolving needs of the user.
- Corrective MA Any change which is necessitated by actual faults (induced or residual) in a software system.
- Risk The potential for realization of unwanted, negative consequences of an event.
- Software Supportability Risk The probability at a given point during the software support phase that the software maintenance activity specified by a baseline software supportability profile can not be accomplished with the available software support resources.

Date:	:	Site:Source of Data:	Page :
		1. S/W BACKGROUND DATA	
		(Complete for each S/W System)	
1 I	DENTIE	FICATION:	
		System:	
1	1.2	S/W System:	
1	1.1.3	S/W System Type (OFP,C3I,EW,Simulator,Missile,ATE):	
1.2 5	DESCRIA	PTION:	
1	1.2.1	Size(#CSCIs, #Modules, #Source Lines):	
1	.2.2	List Documentation Delivered by Contractor and/or	
		Developed During Maintenance:	.
1	2.3	Language(s) & %Use:	
<u>.</u>	1.2.4	Development Contractor Data:	
		Name(s):	
		Development period: Personnel Time :	
1	.2.5	Description of any Major Life Cycle Events(contractor change	e na lor
		modification,etc.):	
_			
¥	2.6	Personnel Currently Supporting S/W System:	
		Total number: List the number by skill level (1 to 5) with 1 = Low, and 5	
		#Lev1 =; #Lev2 =; #Lev4 =; #Lev5 =	
		Indicate approximate % of the time these personnel are dedi	 .cated to
		support of this 5/W system:	
1	2.7	Computer Systems Currently Supporting S/W Maintenance:	
		List computers/peripherals/ or a document containing info	
		Tadicate approximate was the bill black quetons are dedicate	, .+od +o
		Indicate approximate % of the time these systems are dedicated that S/W system:	.586 60
•	2.8	<pre>support of this S/W system:</pre>	
•		List any significant problems which affect this system's so	oftware
		supportability.	

				*	
Date:	Site: S	ource of Data	:	Pag	ge 2
	2. S/W	ASSESSMENT DA	TA		
	•	e S/W System uate : adequ	ate ! <-		
	-50	0	50		
<u> </u>	not_enter_O.	At D	elivery	Current	
	5/W Documentation 2.1.1.1 Modularity : 2.1.1.2 Descriptiveness: 2.1.1.3 Consistency : 2.1.1.4 Simplicity : 2.1.1.5 Expandability : 2.1.1.6 Instrumentation: 5/W Source Code 2.1.2.1 Modularity : 2.1.2.2 Descriptiveness: 2.1.2.3 Consistency : 2.1.2.4 Simplicity : 2.1.2.5 Expandability : 2.1.2.6 Instrumentation:	:			
2.1.3	General S/W Maintainabil	ity :			
On a s	, , , , , , , , , , , , , , , , , , ,	e 5/W Support wate ! adequa v	ite ! <-)		
A	-50	0	50		
<u>Do_n</u>	not_enter_O.	At D	alivery	Current	

<u>Do_n</u>	ot_enter_0.	At	Delivery	Current
2.2.1		:		
2.2.2	S/W Support Systems 2.2.2.1 Host Computer : 2.2.2.2 Software Bench : 2.2.2.3 Lab-Integ. Test: 2.2.2.4 Operational Sys: 2.2.2.5 Other (Specify):	:		
2.2.3	5/W Support Facility 2.2.3.1 Office Space : 2.2.3.2 System Environ.:	:	 	
2.2.4	General S/W Support Envi	ronment:		

Date	·:	Sit	θ:		Source	of D	ata:			P 1 90	3
2.3			-50 to	> ! inad	the 5/W equate	Life ! ad	Cycle S	! <- b	Managemen est	t:	
				-50		•		•			
	<u>Do_n</u>	<u>t enter</u>	_0.			A	t Delive	əry	Current		
	2.3.1	2.3.1.2	Status Config	ion Mgmt fication Account . Contro	: 1:		- - -			-	
	2.3.2	2.3.2.3 2.3.2.3 2.3.2.4 2.3.2.5	Planni Organi Organi Cosign Coding Test	e Mgmt ng zation Methods Methods Methods nterface			- - -			-	
	2.3.3	General	. SWLC S	upport M	anageme	nt: _				-	
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		At Delivery	Current
2.5.1	5/W Supportability Risk	:	
		·	

Date:	Site:	Source	of	Data:	Page	4
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3. DESIRABLE MAINTENANCE DATA FOR EACH SOFTWARE SYSTEM

- .1 FOR EACH BLOCK RELEASE SINCE SOFTWARE DELIVERY
 - a. List of specific software changes implemented
 - b. Estimated person (configuration management, maintenance project) effort
 - c. Actual person effort
 - d. Engineering Start and End dates
 - e. Time from Engineering End date till release was fielded

3.2 FOR EACH SOFTWARE CHANGE REQUEST SINCE SOFTWARE DELIVERY

- a. Id and description
- b. Type (correction, enhancement, conversion)
- c. Priority (emergency, urgent, normal)
- d. Complexity (high, medium, low)
- e. Estimated person (configuration management, maintenance project) effort
- f. Actual person effort
- g. Configuration management open and close dates
- h. Release in which change is or will be implemented

3.3 FOR EACH YEAR SINCE SOFTWARE DELIVERY

- a. Number of software change requests carried over from previous year
- b. Number of software change requests opened during current year
- c. Number of software change requests closed during current year

3.4 ADDITIONAL DATA OF INTEREST

- a. Computer system resources (e.g., computer hours) used for each release
- b. Specific tradeoff factors which were required for each release such as request priority, personnel availability and experience, computer systems availability and adequacy
- c. Major problems which led to delay or inefficiency in completion of a release

D. System Data

APPENDIX D

SYSTEM DATA

D.1 INTRODUCTION.

- a. This appendix contains summaries of the raw maintenance support data gathered from the various sites visited. The sites visited include:
 - (1) NORAD Space Command, Colorado Springs, CO
 - (2) Warner Robins ALC, Robins AFB, GA
 - (3) Sacramento ALC, Sacramento, CA
 - (4) Castle AFB, CA
 - (5) Ogden ALC, Ogden, UT
 - (6) Oklahoma City ALC, Oklahoma City, OK
 - (7) Langley AFB, VA.

At each site, maintenance support data for several systems were collected. Each system (e.g., F-16 at Ogden ALC) generally had several software systems (e.g., FCC, SMS, RDR, HUD). For each software system the maintenance support data consisted of background data, evaluation data, and maintenance activity data on each block release since the beginning of formal software system support activity at the site.

b. The terminology developed to describe the data in a consistent way across software systems is described in section D.2 of this

appendix and appendix B, Glossary of Terms. The actual data collected, and in some cases interpreted from notes and application of the terminology constraints, are summarized in section 0.3 of this appendix. The bulk of the analysis results presented in this report is derived from the data presented in this appendix.

D.2 TERMINOLOGY.

The Glossary of Terms, appendix 3, contains reasonably concise definitions for the terms used in this report. However, there are some caveats relative to the manner in which the actual data is "molded" into the appropriately defined terms. This section is a brief attempt to describe those caveats for the specific data items used in section 0.3 of this appendix.

D.2.1 Background Data.

- a. There were considerable background data collected during the individual interview sessions and from the data survey forms. The more important background data (across software systems) are summarized by software system in section 0.3. The Program Management Responsibility Transfer (PMRT) date is officially when organic software system support is supposed to begin. Many systems have not undergone PMRT, but some have already begun the software support function. In this case, the "delivery" date reflects this unofficial beginning of the support function.
- b. In some cases the organic support is a combination of several organic organizations and/or perhaps a contractor. As much as possible, the personnel counts reflect actual maintenance support personnel, not the personnel which may be part of an ALC overhead management function or a contractor function required in order to process an "official" release, because the software system has not officially undergone PMRT. These actual software maintenance

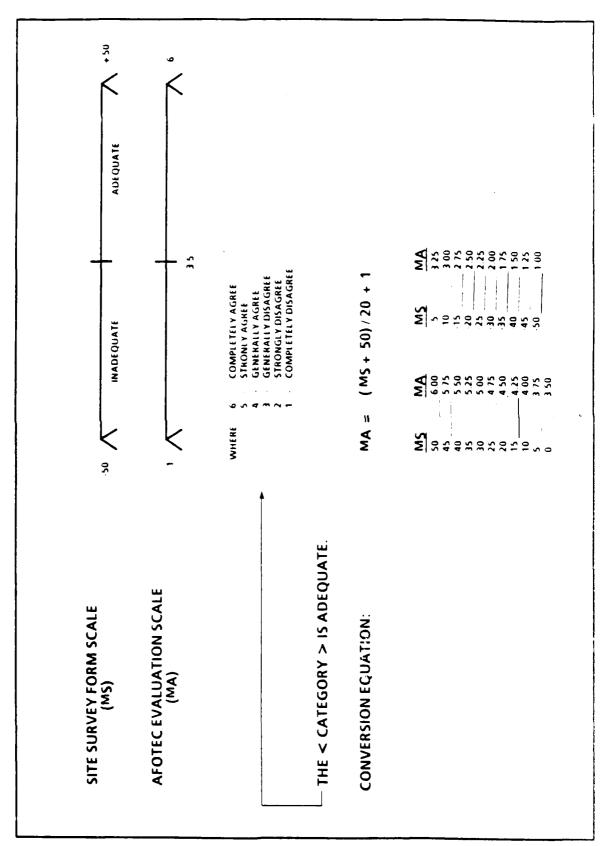
personnel are the management, technical, support, and contractor personnel directly involved in the configuration management and/or analysis, design, code, test of changes in a block release.

c. Major problem areas reflect the particular opinions of the personnel interviewed, and may not reflect the opinion of other management on-site personnel.

D.2.2 Evaluation Data.

- a. The evaluation data for each software system represented the subjective opinion of the personnel completing each data survey form. Each value represented the adequacy of the evaluated supportability category on a scale from -50 (totally inadequate) to +50 (totally adequate). The "O" value is the separator of "inadequate" and "adequate". The transformation of values to AFOTEC's evaluation scale 1 to 6 is illustrated in table D-1.
- b. As an example of how to use table D-1, suppose the evaluated score of the category S/W source code modularity is a 20. Then the corresponding AFOTEC score would be 4.5. This score (4.5) would correspond approximately to a value midway between "generally agree" and "strongly agree" values for the statement: "The modularity of the source code is adequate."
- c. The primary concern of the evaluation was to determine how the supportability metrics compare (correlate) with the concept of supportability risk. Unfortunately, the explanation of risk seemed to be misinterpreted by many evaluation personnel. This has led to a more precise statement of supportability risk as defined in the glossary of terms. As applied to the site survey baselines, the supportability risk is "the probability that the specified block release cannot be accomplished within the available software support resources." "At delivery" the block release would be the first block

Table U-1. Survey Form Evaluation Score Conversion.



release. For "current" the block release would be the current one being processed or, if none were being processed, the next expected block release.

- d. The focus of supportability risk is upon the risk to complete the <u>agreed upon</u> changes in a block release as opposed to all <u>submitted</u> change requests. This focus is required because very little data exist concerning submitted change requests, except for the consensus that there will "always" be more change requests than could possibly be processed. The backlog estimates range from 20 to 200 percent at the current release change count.
- e. Thus, the supportability risk being estimated is the risk of being unable to complete a block release once the contents of a block release have been essentially agreed upon during preliminary analysis. "Unable to complete" is still a fuzzy term, but it includes such things as changes being added and/or additional resources (such as personnel, calendar time, support tools) being required. If the user or any other personnel changes the scope of the block release content in such a manner that the block release will be late or more resources must be added to keep the schedule, then the original block release was not completed as agreed upon. The possibility of this nappening is the supportability risk.
- f. It is clearly realized that there is more to supportability risk than is being measured by this data. However, it does appear that the concept of baseline maintenance support activity (changes in a block release), supportability factors (software products, software support environment, software life cycle management), supportability factor metrics, and supportability risk (as defined here) are reasonably consistent and related terms.

D.2.3 Maintenance Activity Data.

- a. Maintenance activity data are the set of all productivity information concerning each block release of changes to a software system. The data upon which this report focus include:
 - (1) Release start and engineering completion dates
 - (2) Number of personnel available for direct support of the block release
 - (3) Percentage of time these personnel are dedicated to this software system
 - (4) Personnel overlap factor with other releases
 - (5) Number of changes in release
 - (6) Number of changes by type (correction, enhancement, conversion)
 - (7) Number of changes by complexity (low, medium, high)
 - (8) Number of changes by priority (normal, urgent, emergency).
- b. The release start date is that date when analysis activity related to the subject block release begins for which support personnel are required. Typically, this might be the date of the first change request or perhaps the date when no more change requests are accepted for consideration. The engineering completion date is that date when the engineering (including operational testing) part of the block release is complete. Time for "kit" proofing, prom burning, and creation of technical orders after completion of engineering is

<u>not</u> included. There is usually <u>additional</u> time between the engineering completion date, and the actual fielded date. In fact, an engineering release may never be fielded.

- in some direct capacity to the support of the software system. It could be management, technical, support (technicians, librarian clerks), or contractor personnel. The percentage of time these people are dedicated to this software system as opposed to other software systems is required in order to determine "full time equivalent" personnel available to support the software system. This percentage dedicated does include time spent by these personnel performing various "overhead" functions even if not directly related to the software system. Thus, full time equivalent personnel time does include certain overhead time not directly devoted to software maintenance activity. Such time would include:
 - (1) Vacations and sick leave
 - (2) Supporting outside interests such as test agencies and user meetings
 - (3) Support of internal site functions such as internal meetings, and organization training.
- d. In addition to the available "full time equivalent" personnel, it is necessary to account for any overlap by the same personnel in supporting consecutive releases. If consecutive releases involve no overlap, then this factor is 1.0. If 50 percent of the time is spent on each of the releases, then an overlap factor of 0.5 is used for each release. In general, if the release dates (start and end) overlap for consecutive releases, then an overlap factor of 0.5 for the duration of the overlapped time has been used. In other cases, time may have been spent against a planned release which is not completed

and thus never shows up as an overlap. This is totally subjective and can only be accurately specified by the personnel familiar with the given release. Updates to the specified overlap factor (as well as any other data) will be solicited through a normal delphi technique with the software personnel who were the primary source for this data.

- e. The total number of changes in each release is very accurate (except in a few rather obvious instances), and does represent the number of official, documented, change requests (MIP, SPRs, DRs, SMRs, and so forth, as appropriately named by the system's configuration control procedures). The change request generally initiated individual analysis, design, code, and test as well as integrated block release analysis and test. The resulting changes to the software system might be to one module or might be to many modules. The changes to documentation and source code might involve everything from changes to requirements, to simple one-line parameter updates. Just because a change involves only one module does not necessarily imply it is simple. The nature of the change (e.g., development of a state-of-the-art EW algorithm) might dictate much analysis and design, but little code change. The complexity of the change is subjectively defined in terms of scale values high, medium, low in accordance with the combination of skill level of resources required, amount of software product affected, and amount of resources (personnel and support system) required by the change request.
- f. Generally, the number of conversions was not delineated from number of enhancements in the data. Although it was clear from the interviews that <u>much</u> conversion activity is being done, the conversions are usually included with enhancements and are not easily separable.
- g. Except for NORAD, the other sites (primarily ALCs) had only NORMAL (i.e., routine) priority assigned to the change requests.

Concern among ALC personnel for possible problems in adequate response to non-normal priority change requests was indicated. The issue of processing security sensitive changes which fall outside of the "normal" request priority was also raised several times. An estimate of 2 to $2\ 1/2$ times normal change processing time was given for sensitive changes.

D.3 SURVEY DATA BY SITE.

- a. The raw survey data are summarized in this section by site. For each site, the data for each software system consist of background data, evaluation data, and maintenance activity data.
- b. Table 9-2 contains a list of the sites and software systems for which data are included along with the application type of the software system. This table has entries for 81 separate software systems.
- c. Table D-3 contains the software systems background raw data and corresponds roughly to the information requested in section 1.2 of the site survey form (appendix C). These data are reported in six parts. Part 1 is a summary of the data on size in terms of Computer Software Configuration Items (CSCIs), modules, and number of source lines in thousands (k). Part 2 is a list of the primary, secondary, tertiary, and other programming languages in which the software system is written. Approximate percent of source is listed for each language. The dominant language is clearly assembler. Part 3 is a list of system development data in the form of development contractor, development period, and person years of effort. Most of the calendar and effort data in part 3 are approximate. Part 4 is a

summary of the number of personnel assigned to the system, an approximate skill rating from I (low) to 5 (high), and an approximate percentage of the time the assigned personnel are dedicated to the subject software system as opposed to another software system. The skill level generally reflected a level of experience with the subject software. Part 5 is a partial list of the support systems for the software and the percentage of time the support systems are dedicated to the subject software. Part 6 lists software supportability problems reported by the support personnel interviewed during the survey visit.

- d. Table D-4 contains the software supportability evaluation data and corresponds to the information requested in section 2 of the site survey form (appendix C). This table is separated into six parts. The first three parts correspond to the software product, software support facility and software support life cycle management evaluation data for the "AT DELIVERY" system. The latter three parts correspond to the similar evaluations for the "CURRENT" system. data values of -99 indicate data are missing. Raw data values of 99 indicate the category was not applicable for the subject system. All categories in the software product and life cycle management evaluations are applicable. Only a few in the software support facility evaluation (e.g., contractor personnel, "other" support system, and perhaps one of the support system environments) are possibly not applicable. There may be a few typographical errors in the data as to use of the 99 and -99 values, but all other data have been validated against the information entered on the site survey form. Note that there is generally one evaluation per software system. For the F-4 software systems, multiple evaluations were done. This will be helpful for future analysis efforts.
- e. Table D-5 contains the software maintenance activity data as reduced for commonality across software systems. Some of these data are most subjective and need to be reviewed carefully by the cognizant support personnel to improve accuracy.

- f. In particular, the various counts (total, type, complexity, priority) are reasonably accurate if they exist. A zero for all fields of type, complexity or priority indicates missing data. These data would be very helpful if they could be obtained.
- g. The release start date and engineering completion date correspond to the release duration in months. Many of these dates are best guesses. An improvement in accuracy would be a <u>major</u> improvement in computation of the profile charts.
- h. The number of personnel and the percentage of time dedicated to the software system are essentially directly from the background data (section 1.2.6 in the site survey form). Occasionally, these data were missing or conflicted with information obtained during an interview. In these cases a best guess was attempted. An improvement in accuracy of these data would be a <u>major</u> improvement in computation of the profile charts.
- i. The percent dedicated to the release is doubly subjective because it depends upon the accuracy of the release overlaps, <u>and</u> the assumption that, given an overlap, the sharing of personnel is distributed evenly (for <u>each</u> release across an overlap time period). In addition, some of the "quicky" interim/urgent/emergency releases were difficult to categorize. The intent of this percent factor was to reduce, in a reasonably logical and consistent manner, the person time allocated against a given release when the same personnel were being used across several releases for the software. As an example, the NORAD software system releases were overlapped at least three to a year over approximately 11-month release cycles. Any better estimate for this factor would also be a <u>major</u> improvement in computation of the profile charts.
- j. The data in tables D-3, D-4, and D-5 represent a wide variety of interesting information. The current analysis is based upon these

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data. The future analysis to be included in the final draft report will be based upon these data along with as much improved data as can be solicited from the original evaluators and support personnel interviewed.

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SYSTEM	SOFTWAKE SYSTEM	SUPPORT COMPUTER SYSTEMS	DEDICATED
	140	HARKIS / S, C/R, 1/F, PAFER 14FE	ot
		REHUER/FLOWIN, DISC P/E B/32, L/P, MI 300MB DISC, FLOPPY, FAPEN IGEN GADER/PHACH)
+ C- 1 35	- 'S' 3	F/E 8/32, L/P. 80 MB DISC	٥
I-4 TRAINER	1-4 SIMULATOR	GA 16/440, DISC, L/P, FLOPEY DISC, C/R, TERMINAL PAPER TAPE KEADER/PINKTH	100
) + CC	IBM 4341, DEC 10, FDF 11s, ZENITH 100, IBM FC	OB.
	9	CORE ALSO TROT, CAIST, MAD CONCRETE CORY 11/2%	Ŧ
	110	NORSK DATA COMPUTER, PERKIN ELMEK, 2	001
		TERMINALS, VERSATEC PRINIEH	,
	FCR	VAX 11/750, DEC 10, RAINBOW 100	Ö n
	SEG	DEL 19, FUR 116, MAINBUM 199, (BEE ALSO FEBY, CRISF, AND OS/CMF	Q a
	MDIS	MDIS S/W DEV. SYSTEM, MDIS FIELD SYSTEM, SIATIC SIMULATOR TEST STAND, DYNAMIC TEST STAND,	OB
	MDIS	MUTS S/W DEV. SYSTEM, MDIS FIELD SYSTEM, STATIC SIMULATOR FEST STAND, DYNAMIC TEST	OЯ
		STAND	
	AN/AKN-101	DEC VAX 785, PDF 11/34, IBM 5083, SEL 52/75, HE 1000	69
	AN/ARN-101	IN 1003 (4 TERMINALS), VAX 11/785 (UNLIMITED	30
	AN/ARN-101	DEC 1487 785, PDP 11/34, IBM 3483, SEL 52/75,	9
	ANZAKN-101	HE 1000 DEC VAX 285, FDP 11/34, IBM 3083, SEL 32/75,	4
	BALENN-101	HP 1000 DEC WAX 785, PDP 11/34, 16M 3083, SEL 32/75,	4. 4)
	ANZARN-101	DEC VAX 785, PDP 11/34, 18M 3/83, SEL 31/75,	4
	ANZAKN-191	MF 1040 VAX 782, VAX /85, IBM 5083, FDF 11/34, FDF	10
	AN/AKN-101	11/60, SEL VAX 782, VAX 785, 18M 3083, FDF 11/34, FDF	2
	ANZARN-101	11/60, SEL VAX 782, VAX 785, IEM 3083, PDF 11/34, 1DF	
	901-080×00	11760, SEL VAX 782, VAX 785, 18M 3083, PDP 11734, FDF	3
	AN/ ARN-101		٤
	1 KU-1 / HCM	1160, SEL 32/75,	<u> </u>
	1.KU 1740M	FOF 1160, SEL 327/5, VAX FOF 1160, SEL 327/5, VAX	<u>.</u>
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30		MINUTEMAN 11	WING VI/HEG/KAIS	IBM 4341	3
ŝ		KF-4C	HNZHKN- 101	DEC VAX 785, PDF 11/54, IBM 3083, SEL, FDF	69
				11/60, VAx 782	
.) 10	מת - או ר	ÆF : 4ι.	AN/AKN-101	IBM 3083 (4 TERMINALS), VAX 11/785 (IML MITED TERMINALS)	07
3,	OD ALC	KF -4C	AN/AKN-101	DEC VAX 785, PDF 11/54, 18M 5083, SEL 52/75,	45
				HF 1000	
3	OO 101	K.F 4 €	ANZAKN-POL	DEC VAX 785, PDP 11/34, 18M 3083, SEL, PDP	06
4	3 1 00	7. H. 4.	AN/AKN-101	11/60, VAX /82 DEC VAX /85, FDF 11/34, IHM 3083, SEL, FDF	ě
,				11760, VAX 782	
0	00 ALC	hf 4C	GN/ARN-101	DEC VAX 785, PDP 11754, IBM 3083, SEL, 1 DP	, ,
				11/60, VAX 782	
٥		<u> </u>		16M 4341, ELECTRONIC SYSTEMS 1651 SET (ESTS)	<u>.</u>
2	Ut HI L	ארנה		IBM 4541, ELELIKONIL BYBIEMB (EB) BEI (EB)S, IEGNIS OIG EDIT GAELHADE CIATUM (1666)	<u>}</u>
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9	UK-ALC	HLCH	()FP	IBM 4341, INSTRUCTION LEVEL SIMULATOR,	2
				SUBSYSTEM SIMULATOR	
6		B−1B ·	LADC	VAX 11/780, 18M 4341	-
Ð	_	H-16	S113	VAX 11/780, IBM 4341	-
0		n .	EMUX	VAX 11/780, 18M 4041	-
Ξ;	_	H1-H	F/LGMS	CAX 11/780, IBM 4641	
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1.4	3		SMCF	IEM 4341, E-3 AISF	_
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9		EL.CM	5-10	KULM 1666B, ENHANCED DISK PRODUCTION SYSTEM,	à
				2 DUAL FASS DRIVES, DATA BENERAL MUTORINA	
ם		FILLOW	A · D1D	RUM MIGGED, RASS DRIVE	ij
D 4	O C.	£3 13	Ĭ.	KULM 16668, DUAL KASS DRIVE, ZEBKA DISP	•
				DRIVE, SOFTWAKE DEVELUPMENT SYNTEM	
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			<pre>fable 0-5. Systems background Raw Part5: SUPFUKI SYSTEMS(FAKTIAL)</pre>	fable 0-3. Systems Background Raw Data Parts: SUFFUKI SYSTEMS(FAKTIAL)	
10	SITE	SYSTEM	SUFTWAKE SYSTEM	SUFFORT COMPUTER BYSTEMS	2 TIME DEDICATED
B	- INET HIB	E-3A	AUCF	IBM 370/168, 3 SPECIAL FUKPOSE IBM-4FI SIMU ATOKS	97) 97)
'n	III EN AFS	E-34	UTILITIES	IBM 370/168, 3 SPECIAL -PURPUSE IBM-4FI	ın
ž	LANGLEY	J1105	AS11/1F00F	BINALHIUND IBM 4341 WITH 4 MEG. MAIN MEMOKY IN IM ULUCES ON LINE STORAGE, FRINTER, 3270 TERMINALS, INTERFACE SIMALATOR ANALYZER (ISA) NOVA	2
ž	LANGLEY	STRIS	STRIS	4/x (INCLUDES 25M DISF, PKINIEK, TEKMINAL) IEM 360//O, ALL PEKIPHEKALS, DOS 5250, ALL PEKIPHEKALS, FOR 11/20 TAPE DKIVE	2
÷,	L ANGLEY	I ACS	CAFMS	PE 3230-1251, TRIDENT DISTURE, REMEN TAPE	100
F)	LANGLEY	1.41.1	DC/5R	(1) ANIVE-7 MAINFRAME, (4) LOST KOHM MINI-COMELIFERS (4) TOTO OLSE THE COME	100
*	L.ANGLEY	1411	11/MARRES/TEREC	SEE CRISF & 0/5CMP, 15M 4341 V5/05	301
ş.	LHANGLEY	4137L	HUGHES UTIL	HUGHES 4118	21
9		4071	IBM UTIL	IBM 4341	S.
>	LANGI EY	4071	10KP / 1MFP	H-4118(2), 18M 4341, 40s SIMIRALE, 40s NUI	an an

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2 :	S118	aYSTEM	SOFTWARE SYSTEM	PRUBLEM(S) DESCRIPTION
-	O HOM	95		
• ~	NUKAD	MEBU	MEBU	(1) Lack of tools to locate and debug railures.
M -	NOKAD	NCS 	SIZ	(1) Lack of tools to locate and debug failures.
rin	WK-ALC	ALK-46	ALR-46	(1) Inadequate number of trained softw.re personnel.
				(2) Time to get completed software fielded is much too
0	WR-ALC	ALK-67	ALR-69	iong. (1) Inadequate number of trained software personnel.
7	WK-ALC	AN/AL 0-131	B1G	(1) Inability to adequately staff positions prior to
				ring to ensure motivate quality.
			,	expertise.
œ	#K -ALC	AN/AL U- 1 51	5	(1) Inabijity to adequately start positions prior to FMK1 to ensure software quality.
				C. Indebility to maintain the required sevel of expertise.
3	WK-ALL	AN/AL0-131	וחו	(1) Insufficient trained personnel before and after
				PMKI. (2) Inability to maintain required level of expertise.
Ξ	WK-ALC	HFR 36	APK - 38	
=	WK-HI C	B-52 EVS ATE	AS0-151	(1) Limited time available on test set where software
Ç	MR: DI	F 20 0010NB 5 41E	(H) (H) - H(H) / N(H)	
: ::	WR-ALC	SA HVIONICS	AN/GSM-1885 (W)	
4	EN INC		ככ	
7	EK PLC	F 15	KADAK	(1) inadequate logistical support in ortaining
				engineering services, computers, other hardware, consumables, parts, & LRUs.
				(2) Lack of technical personnel in the areas of radar,
				Offe, computer science, & hardware.
				requirements throughout the AF.
2	WR-ALL	F-15 AVIUNICS AIE	ADTS, A15	(1) Configuration control after PMKT.
				(3) Interface between ALL and control.
_	WA-ALL	JIIDs	AS11/0CP	
ם	WK ALL	31105	E- 3A AWACS/OCP,	 bifficulty obtaining memory boards bifficulty obtaining FkWN.
7	WK-ALC	21115	SF7USEA	
Ξ,	MY - 141 C	31105	SYS EXERCISER	
- .	3 T 3 T 3 T 3 T 3 T 3 T 3 T 3 T 3 T 3 T	FAVE THE	61 SF	
:	WK HLU	211		(1) hist nationals unavailability. (2) interface between AF organizations.
~			77.3	(s) Retention of qualified personnel (i) General software common and confineration
•				menagement suffer considerably from a shortage and
7) H4 H2	ł 111F	WITC	Continual turnover or qualitied engineers. (1) Organic gottware support and continuestion
				management burger Lumphoer and y riom a microsystem of continual turnover of qualified engineers.

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			(abie u-3. Systems Farté: SUPPUNIA	SUFFURIABILITY FRUBLEMS
10	SIJE	SYSTEM	SOFTWAKE SYSTEM	PROBLEM(S) DESCRIPTION
Ą	3M-ALC	FB-111A	M-4C	(1) Organic software support and configuration management suffer considerably from a shortage and continual turnover of qualified engineers.
45	CASTLE	B-52 B-52	EN:	
9 7		1.C-135 1-4 TEMINER	WST 1-4 SIMULATOR	
Ř	OO ALC	F-15		(1) Number of configurations, customers, change
7.		F-16	HUD	drivers, atc.
3		F-16	DF.T	
				 Descriptiveness of documentation. Softwars distribution support.
I)	. Cuu : At L	10	FCR	
				(2) Lack of test bench. (3) Lack of manpower.
				(4) High turnover of personnel.
				(b) Code modification requires knowledge of software,
ь, 4		1-10	SMS	istoners, recer, eld no.
12	OUT ALC	1 - 4	HDTS	(1) Unable to hire sufficient number of technical
				9 1
3.6	00 - ALC	4-4	MDTS	to be addressed during past or present ipdates. (1) Too such time is required to fully inderstand and
				be able to utilize the numerous capabilities of the operating system.
37		F-4E	AN/4KN-1Q1	
6	-	F -4E	HN/ARN-101	
65		F 4E	AN/AKN-101	(1) Insufficient manpower.
÷.		F. 4氏	ANY DEN - 101	
7		7 - 1-4 E	ON CARNATOL	
4 4	00 -4 C	, 7 - 4E	60/500-101 60/500-101	(1) Support system is currently incomplete.
4		1 46	AN/AKN-101	
45	00 -ALC	1-46	AN/AKN-101	
40		f -46	AN/AKN-101	Support
4	00-₽.C	- 46	AN/ARN-101	Support
D T		145 145	LKU-1/ALM	(1) Most of the Work consists of software enhancements that require an encironarion harborized sather than
				that require all mighings ing backyr culto rather than a software background.
44			1 KU-17ACM	
ñ	00- ні.∟	+ 4G	L KU-17ACM	(1) fact of engineers qualified to solve complex
7.	00 000	HINDLEMAN	BTOCKET SNIM	(1) Contention of current projects.
1,		MINUIEMAN	WING VI/HS-29	
ji i			WIN65/H5 - 28	(1) Contention of current projects.
IJ 4	J # (F)	MINULEMAN II	SSAS/LAFS	(1) Ditdaled software system (currently peing inpreded).
33	. O PH C O	MINULFIAN II	WING VAREBARATS	٠
				(1) regard at the Sample of Tables.

Systems background Kaw Data SUPPORTABILITY PROBLEMS	FRUEL EM (S) DESCRIPTION	(1) Inmufficient manpower.	(1) A major problem has been the turnover of our experienced people and their replacement with inexperienced personnel. (2) We have problems establishing a training program using our experienced people, since they are committed to other projects. (3) Maintainability of the subsystem simulator is poor.	 (1) Inadequate documentation. (2) Extremely poor contractor design. (3) High turnover rate. (4) Insurticent management 		1		(1) Inadequate documentation on system design and sortware modules. (2) Inadequate identification and information on support equipment and software. (3) AF/contractor interface. (4) Inability to get necessary information from contractor interface.	(1) The uniqueness of trainer hardware requires that timel development and testing he done on an actual trainer. This complicates development and training schoolies.		 In the Elective MS4 document, over 50% of the document does not march the code or reaching conventions. Ingre above may to organically support the INS program until an inertial computer (INC) is installed in the life cycle support taxibity (ISS).
lable D~3. System Part6s SUFPORT	SOFTWAKE SYSTEM	MING VI/HEG/KATS AN/ARN-101 AN/ARN-101 AN/ARN-101 AN/ARN-101 AN/ARN-101 I I I I I	τ Σ	7-10. CADC	5113	ЕМЈХ	F / CGMS I NS	ÜKS	E N S		MC _ EAEC INS
	SYSTEM	MINUIEMAN 11 NF-4C NF-4C NF-4C NF-4C NF-4C NF-4C AF-4C A-CM	E O ਵ	71. 74 14	h-1 Б	b-1b	B - 18	ម . ច	4 G H	74 A.A.A.A.A.A.A.A.A.A.A.A.A.A.A.A.A.A.A.	ମ ଫ ମ ୟ ଧ
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	SUFTWARE SYSTEM SUFTWARE SYSTEM UNDEGA SMCF SKGF SKGF SKGF SKGF SKGF SKGF SKGF SKG	. D-3. Systems Background Raw Data Parto: SUPFORTABILITY PROBLEMS	EM PHOBLEM(S) DESCRIPTION	(1) NDC 1070 has limited memory. (1) Software configuration management controls. (2) Equipment maintenance problems caused by supply system. (3) Lack of AISF engineering and configuration.	Selegement for natoward to the Demands. (1) Subgrantial Overhead work demands. (2) Demands in the American Fig. 514.		computer. (1) Authorized manpower is inadequate. (2) Support equipment is also required for product.	production. (1) Several programs exist which are now being integrated into one consistent paclage. Without integration, supportability would be seen a difficult, the second is programs and consists.	Colline and the second and the state of the second and the second		(1) Complete dependence on the contractor to support	software.	(3) Inmenforced standards. (3) Foor acquisition standards. (4) No coding standards.		 (4) Insurtationary representation don'ing motivary design. (5) Lack of HIGHES class I terminal(8). (5) Lack of militarized operator interface units(OIUs). (4) No maintenance contract for ISH. (5) Generally disjointed program management (ESD, IEM, WANNER KOBINS ALC, etc.). 	(1) Graphic software is maintained in Pid(KU II) assembler on non supported operating system. Limited number of howledgeable operators. (1) US SASO Editor is a line editor and only I terminal (a) no used at a time.	The training time is 1.
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Table 0.5. Systems Background Raw Data Parts: SUFFURIABILITY PROBLEMS	PKOBLEM(S) DESCRIPTION	(1) No AIC training support in MAP 18M sesembler language. (2) Significant down time on 4118 because of age and lact of spare parts. (3) Significant amount of time dedicated to interface testing.
Table D.S. Systems Fartos SUFFURIA	SOF TWAKE SYSTEM	LBM OFFIL. TOKEZIMER
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	115 - G118	76 LANGLEY

Fage No.

Lable D 4. Systems Evaluation Raw Data
Parti: SUFTWAKE PRODUCT MAINTAINABILITY (AT DELIVERY)
Scale: - Section to + 50(High) i - 99(Missing) i + 99(N/A)

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fable D-4. Systems Evaluation Raw Data Fartii SOFTWAKE FRODUCT MAINTAINABILLITY (AT DELIVEHY)

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Table D-4. Systems Evaluation Raw Data Part2: SOFTWAKE SUFFORT FACILITY (AT DELIVERY)

Scale: - 50(Low) to + 50(High) i - 99(Hissing) i + 99(N/A)

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Table D-4. Systems Evaluation Kam Data Fart2: SUFTWAKE SUPPORT FACILITY (AT DELIVERY)	χ + υ	PER	6 2 N 2 G
Table D- Fart2: 50FTW	Scale: - 50(Low) to + 50(High) i - 99(Missing) i + 99(N/A)	SOF TWAKE SYSTEM	DC/SR 11/MAKKES/TEKEC HUGHES UTIL 16M UTIL 10KP/1MPP
		SYSTEM	11F1 11F1 467L 467L 467L
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Part 3: LIFE CRILE SUFTWAKE SUPPORT MANAGEMENT (AT DELIVERY)	10/19/00 4 4 (00/19/19/19/19/19/19/19/19/19/19/19/19/19/
	Fart3: LIFE CRILE SUFFINAKE SUPPORT MANAGEMENT (AT DELIVERY)

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Table D 4. Systems Evaluation Raw Data Fart St. LIFE CYCLE SOFTWAKE SUPPORT MANAGEMENT (AT DELIVERY) Scale: - 50/Low) to + 50/High) t = 99(Missing) t + 99/M/A)

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Table D.4. Systems Evaluation Raw Data Part3: LIFE CYCLE SOFTWARE SUPFORT MANAGEMENT (AT DELIVERY) Scale: - Sociow) to + SO(High) : - 99(Missing) : + 99(N/A)	SOFTWAKE SYSTEM	DC/SR 11/MAKKES/TEKEC HUGHES UTIL 16M UTIL 10KP/TMFP
	SYSTEM	11P1 11P1 11P1 407L 407L
09/26/85	ID SITE	93 LANGLEY 94 LANGLEY 95 LANGLEY 96 LANGLEY

Table D-4, Systems Evaluation Kaw Data Part4: SOFTWAME PRODUCT MAINTAINABILITY (CURRENT)

1997 Se. 85

Scale: - 50(Low) to + 50(High) i - 99(Missing) i + 99(N/A)

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SVSTEM	CSS NEED NCS SSC ALK-69 AN ALG-131 AN ALG-13
ID SITE	1 NORAD 3 NORAD 4 NORAD 5 NORAD 5 NORAD 5 NORAD 6 NW - ALC 7 NW - ALC 7 NW - ALC 11 WW - ALC 11 WW - ALC 11 WW - ALC 12 WW - ALC 13 WW - ALC 14 WW - ALC 15 WW - ALC 16 WW - ALC 17 WW - ALC 18 WW - ALC 19 WW - ALC 10 WW - ALC 10 WW - ALC 10 WW - ALC 10 WW - A

Table D.4. Systems Evaluation Raw Data Part4: SOFTWARE PRODUCT MAINTAINABILITY (CURRENT)

		Scales - 50(Low) to	• 50	50 (H1 gh)	-	99 (M1	99 (Missing)	•	99 (N/A)	(A)							
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58 00-ALC	KF - 4C	AN/AKN: 1::1	ε	35	S		25		40	_						33	10
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Table D-4. Systems Evaluation Haw Data Part5: SOFIWARE SUPFORT FACILITY (CURRENT)

- 50 (LOW) to + 50 (High) i = 99 (Missing) i + 99 (N/A)

Scales

Fage No.

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Raw Data	SOFTWARE SUPPORT FACILITY (CURRENT)
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SOFTWARE SYSTEM	AN/ARN-101 LRU-1/ACH LRU-1/ACH LRU-1/ACH LRU-1/ACH MING 11/2015 WING 11/2015 WING V/HS-28 SSAS/CAFS WING V/HEG/RATS WING V/HEG/RATS AN/ARN-101	CAFMS
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10 SITE	47 00-42 58 00-42 58 00-42 58 00-42 58 00-42 58 00-42 58 00-42 58 00-42 59 00-42 50 00-	92 1 ANG

58/97/		Table D- Part5: SOF	0-4. Bystems Evaluation SOFTWARE SUPPORT FACILITY	tens	Evalu	ation ILITY	Raw (CUR	Raw Data (CURRENT)									
		Scale: - Su(tow) to + 50(High) i - 99(Missing) i + 99(N/A)	0 + 20	H1 9h)		1W) 66	ssing	-	N) 66	(A)							
10 S17E	SYSTEM	SOF TWARE SYSTEM	PER	PER MAN	PER TEC	AE SUF	PER P	SYS	SYS HOS	AE SYS BEN	AE SYS LAB	AE SYS OFE	AE SYS OTH	A 5	PAC DFF	P. P. P. E. P. C.	ENC- IRON
	4	98/52	9	20	-10	9	01	30	2	66	Ş	8	66	50	10	30	20
94 LANGLEY 95 LANGLEY 95 LANGLEY 97 LANGLEY	11P1 467L 467L 407L	INARRES/TEKEC HUGHES UTIL IBM UTIL IORP/IMPP	. 1 1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	18 to 0.83	2222	22-2	3	10 20 20 40	25 16 20 40 40 40	2992	3 2 4 5 5 5	0 6 6 6	949	20 07 40 70 40 70	20 -20 35	20 10 45	20 m m 0

Table D-4. Systems Evaluation Kaw Data Fartsi LIFE CYCLE SOFTWAKE SUPPORT MANAGEMENT (CURKENT)

Scale: -50(Low) to +50(High) : -99(Missing) : +99(N/A)

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3	AN/ALG-1.1	B TG	_	-20						_	_	_			2	50	0.30
ĭ	AN/ALD-131	0FP		-20						္ဌာ	01	_			9	50	0.30
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Table 0-4. Systems Evaluation Raw Data Parto: LIFE CYCLE SUFTWARE SUPPORT MANAGEMENT (CURRENT) Scale: - Solicon: to + Sol(High): 1 - 99(Missing): 1 + 99(N/A)

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Table D-4. Systems Evaluation Raw Data Parts: LIFE CYCLE SOFTWARE SUPPORT MANAGEMENT (CURRENT)	Scale: $-50 (Low)$ to $+50 (High)$: $-99 (Missing)$: $+99 (N/A)$		SOFTWARE SYSTEM	!!!!!!!!!!!!!!!!!!!!!!!!!!!!!!!!!!!!!!!	DC/SR	II/MARKES/TEREC	HUGHES UTIL	IBM UTIL	10RF/IMPP
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E. Systems Descriptions

APPENDIX E

SYSTEMS DESCRIPTIONS

Because it is highly probable that all the software systems reviewed in this study will not be familiar to the reader, this appendix contains systems descriptions for most systems listed in table E-I. For some systems, appropriate descriptions were not available to the authors.

Table E-1
Software Systems Examined

SITE NORAD NORAD NORAD NORAD NORAD WR-ALC WR-ALC WR-ALC WR-ALC WR-ALC	SYSTEM CSS MDS MEBU NCS SSC ALR-46 ALR-69 AN/ALQ-131 AN/ALQ-131	SOFTWARE SYSTEM CSS MDS MEBU NCS SSC ALR-46 ALR-69 AGEOP BTG
WR-ALC WR-ALC WR-ALC WR-ALC WR-ALC WR-ALC WR-ALC	AN/ALQ-131 ALQ-131 APR-38 B-52 EVS ATE E-3A AVIONICS ATE E-3A AVIONICS ATE F-15 F-15	OFP UUT APR-38 ASQ-151 AN/GSM-285(B) AN/GSM-285(W) CC RADAR
WR-ALC WR-ALC WR-ALC WR-ALC WR-ALC WR-ALC SM-ALC SM-ALC SM-ALC CASTLE AFB CASTLE AFB CASTLE AFB CASTLE AFB CASTLE AFB CASTLE OO-ALC OO-ALC OO-ALC OO-ALC OO-ALC OO-ALC OO-ALC OO-ALC	F-15 AVIONICS ATE JTIDS JTIDS JTIDS JTIDS JTIDS PAVE TACK PAVE TACK F-111D F-111F FB-111A A T-4 B-52 B-52 KC-135 F-16 F-16 F-16 F-16 F-16 F-16 F-16 F-16	ADTS, AIS ASIT/OCP E-3A AWACS/OCP SP/USER SYS EXERCISER AISF OFP WEAP-NAV COMPUTER WEAP-NAV COMPUTER WEAP-NAV COMPUTER A T-4 SIMULATOR CPT WST WST FCC HUD OFT FCR SMS MDTS AN/ARN-101 AN/ARN-101
00-ALC 00-ALC 00-ALC 00-ALC	F-4G MINUTEMAN MINUTEMAN MINUTEMAN	LRU-1/ACM WING II/2015 WING VI/HS-29 WINGS/HS-28

Table E-1
Software Systems Examined (Continued)

CO-ALC OO-ALC OC-ALC	MINUTEMAN II MINUTEMAN II MINUTEMAN II RF-4 ALCM ALCM ALCM ALCM B-1B B-1B B-1B B-1B B-1B B-22 B-52 B-52 B-52 B-52 B-52 B-52 B-5	SSAS/CAPS WING V/HEG/RATS WING VI/HEG/RATS CAN/ARN-101 LEVEL 1 TEST LOADED PYLON TEST OFP CADC CITS EMUX F/CGMS INS ORS BNST FTSS MC-1 EXEC MC-2 EXEC AINS OMEGA SMCP SRCP SRGSCP DPS M-DTD MPT OFP WCS OFP AOCP UTIL SUPP S/W ASIT/TPOCP STRTS CAFMS DC/SR II/MARRES/TEREC HUGHES UTIL IBM UTIL
LANGLEY	407L	IORP/IMPP

THE BDM CORPORATION

ID : 1

SITE : NORAD SYSTEM : DSS

SOFTWARE SUBSYSTEM : CSS

SOFTWARE BUBSYSTEM TYPE : C-E

DESCRIPTION:

The NORAD Chevenne Mountain Complex (NCMC) software is a domplex of communications-electronics systems for space surveillance, missile warning, and related temmenications at support functions. The five major software sous, stems (Nollie NCS, SSC, CSS, MEEU, and MDS.

The Communications System Segment (CSS) provides the required communications interfaces between program elements, between NCMC systems and external systems. It consists on Moneywell Information System (HIS) and NOVA digital computers and application software firmware.

ID : 1 SITE : NORAC SYSTEM : MDS SOFTWARE SUBSYSTEM : MDS SOFTWARE SUBSYSTEM TYPE : 0-E DESCRIFTION:

The NORAD Cheyenne Mountain Complex (NCMC) software is a complex of communications-electronics systems for space surveillance, missile warning, and related communications a comport functions. The five major software subsystems in 11.33 NCS, SSC, CSS, MEBU, and MDS.

The Modular Display Sub-system (MDS) consists of Raviheon consoles, Data General NOVA Digital Computers, Ford Aerospace and Communications Corporation (FACC) developed hardware and joint FACC and System Development Corporation developed software/firmware. It provides the NORAD Computer System (NCS) and the Space Surveillance Center (SSC) with the required man-machine interface with the Embedded Computer Resources in the NCMC. Additionally, a CINCNORAD Remote Display Information Terminal has been established which provides CINCNORAD on-line access to NCS data.

THE BDM CORPORATION

ID : T SITE : NORAD SYSTEM : MEBU SOFTWARE SUBSYSTEM : MEBU SOFTWARE SUBSYSTEM : YEE : C-E DESCRIFTION:

The NDRAD Chevenne Mountain Complex (NEMC) software is a complex of communications-electronics systems for space surveillance, missile warning, and related communications and support functions. The five major software subsystems include NCS, SSC, CSS, MEBU, and MDS.

The Mission Essential Backup/Command Center Processing and Display System (MEBU/CCFDS), consists of UNIVAC 1100 series digital computers and associated display hardware. UNIVAC standard software, MEBU unique software, and CCFLS software. The MEBU provides backup to the NCRAD Computer System (NCS) for the Missile Warning Mission of NCRAD.

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THE BDM CORPORATION

ID : #
SITE : NGRAD
SYSTEM : NGS
SOFTWARE SUBSYSTEM : NGS
SOFTWARE SUBSYSTEM TYPE : G-E
DESCRIPTION:

The NORAD Chevenne Mountain Complex (NOMC) software is a complex of communications-electronics systems for epace surveillance, missife warning, and related communications one support functions. The five major software subsistems the ode NOS. 880, 088, MEBU, and Mub.

The NCRAD Computer System (NCS) consists of the world wide Military Command and Control System (WWMCCS). Honeywell Information Systems (HIS) subjusted computers and associated display hardware, related WWMCCS standard software and applications software. The NCS provides CINCNURAD with the required computer resources for command and control of the NCRAD forces and for missile warning operations.

The NCS Operations programs comprise a major portion of the NCS software. These programs benform the calculations and data manipulations directly associated with aeruspace defense and warning, generate and process simulated data in real-time, and record operational data. They accept real-time inducts from operators and from data sources external to NOFAD and generals real-time outputs for internal display and transmission to be other users.

The MCS Support software performs the calculations and late manipulations directly associated with exercise generation and data reduction. Its inputs are primarily cards, tape files secural files, and it generates table and disp files for use ovi a ferations and utility CFCI and printed reports for use ovi luerations and frogramming personnel.

The NGS obttitive suftwape commentees a minor contion of the colling of subjections and contions of productions are productions and support of the CS Operations and NGS Subject 17017985. It accepts realistance and confinestivities and generates ourseld for internal display thomas contions and generates ourseld for internal display contines. Darameter assembly testing, and lightany courties.

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THE BDM CORPORATION

ID : 5 SITE : NORAD SYSTEM : SSC SOFTWARE SUBSYSTEM : SSC SOFTWARE SUBSYSTEM TYPE : C-E DESCRIPTION:

The NORAD Chevenne Mountain Complex (NCMC) software is a complex of communications-electronics systems for space surveillance, missile warning, and related dominunications and support functions. The five major software suppyshems include NCS, 380, CSS, MEBU, and MES.

The following 350 functional areas are defined in terms of the capabilities required to support that area of the mission.

- (1) Astrodynamic Support provides for accurate, precise, rapid astrogynamic computations.
- .2. Openations Canter Control supports command direct.in and control of SSC processing.
- (3) Automatic Catalog Maintenance satisfies requirements for automatically initiated (data triggered) processing of sensor observations, and correction and transmission of satellite orbital elements.
- (4) Launch Processing provides for detection of new satellite launches, generation and maintenance of proital elements of new satellites, and control and efficient use of the sensor network during the new-launch time Frame.
- (5) Breakup Processing provides for efficient degeracted, correction, maintenance, and cataloging of protest atements an associated with pressups.
- 5. Maneuver Processing provides for detection and analysis of satelline maneuvers, and for generation of elements or new/inanged satelline ordins.
- .7) Sensor Control provides for monisoring sensor environment and performance status, managing required sensor tata pollection, and optimizing use of the sensor retwork.
- By Manual Analysis ond. ides for manually iditiated selection of observations and evaluation of orbital economists.
- PRO Espenhal Daga Anddelts phovides for denamation of some little element datalled lend officered to for dag do little element datalog data daens.
- (\cdot) (i). Special Mission Friguets provides data for use 1 H.3 have logividualized support.
- 11 Umbital Analyst Araduces provides for Mark we what we shall be some events. And for perention of data week to describe and properties of descripting sateriates on the element of descripting.
- (1) Management Endducts subscrits requirements for other.etical reports on the other industrate of bister industrial
 - ್ತಿತ್ರ ತಿಳಿತ್ರವಾಗ ಹೆಚ್ಚಾರು ಕ್ರಾರಾಜಕ ಅವರ ಕ್ರಾರಾಜಕ ಕಟ್ಟಿಯ ಕ್ರಾರಾಜಕ ಕ್ರ

THE BDM CORPORATION

ID : 5 SITE : WR-ALC SYSTEM : ALR-46

SOFTWARE SUBSYSTEM : ALR+46 SOFTWARE SUBSYSTEM TYPE : EW

DESCRIPTION:

ALR-46 is a threat warning system. The threat warning system is a software programmable radar warning set which alerts the pilot to the presence of signals emanating from threat radars. Identity and relative bearing of each threat are presented to the pilot to enable the options of avoidance and/or use of countermeasures.

THE BDM CORPORATION

ID:

SITE : WR-ALC SYSTEM : ALR-59

SOFTWARE SUBSYSTEM : ALR-69 SOFTWARE SUBSYSTEM TYPE : EW

DESCRIPTION:

ALR-09 is a threat warning system. The threat warning system is a software programmable radar warning set which alerts the bilbt to the presence of signals emanating from threat radars. Identity and relative bearing of each threat are presented to the pilot to enable the options of avoidance and/or use of countermeasures.

THE BDM CORPORATION

V.

ID : 8
SITE : WR-ALD
SYSTEM : ANVALUE-1I1
SOFTWARE SUBSYSTEM : AGEOF
SOFTWARE SUBSYSTEM TYPE : EW
DESORIFTION:

The Aerospace Ground Edulpment Operating System (AGEDF) is resident on the AN/ALM-186 and provides initialization of the AN/ALM-186, task scheduler, disk file manager, software onliver for I/O devices, and command interpretation. It also provides the operator interface to SUT software through a test executive program.

THE BDM CORPORATION

V

10: F SITE: WE-ALD SYSTEM : ANYHLE-101 SOFTWARE SUBSYSTEM : BTS SOFTWARE SUBSYSTEM TYPE : EW DESCRIPTION:

The flue Tape Generator (ETG) is an interactive program to generate the data required by the ECM system Operational Filads Frogram (CFF). This data consists of poth mission data (i.e. threat definitions and jamming beconsidues) and aircraft data (i.e. amount of prime power available). The imput to be $\pm 3\%$ is alpha-numeric text and the bulput is binary data caples.

THE BDM CORPORATION

ID : 10 SITE : WRHALD SYSTEM : AN/ALDHID1 SOFTWARE SUBSYSTEM : OFF SOFTWARE SUBSYSTEM TYPE : UFF DESCRIPTION:

The Operational Flight Program (GFF) provides control, fault detection, and hardware resource allocation during flight operations. These functions are based on data which is the output of the Blue Tape Generator and control commands from the cockpit.

THE BOM CORPORATION

ID : 11 SITE : WR-HLD SYSTEM : AN/HLD-171 SCFTWARE SUBSYSTEM : UUT BORTWARE SUBSYSTEM TYPE : ATE DESCRIPTION:

The Unit Under Test (UUT) software is resident on the Admid-136 and provides the interface among the support equipment computer. UUT (Advance-131) and the maintenance technician in the field shop. This software provides some limited automatic test capability and manual test instructions to the technician for system checkout and fault isolation.

THE BDM CORPORATION

ID : 12 SITE : WRHALD SYSTEM : AFRHUB SOFTWARE SUBSYSTEM : AFRHUB SOFTWARE SUBSYSTEM TYPE : EW DESCRIPTION:

The APR-T8 Homing and Warning system is comprised of four subsystems having a total of 25 wine Replaceable Units. The primary subsystems include: Receiver Set. Control Indicator Set. Homing and Warning Computer, and Computing and Optical Sight System.

THE BDM CORPORATION

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E: 17 SITE: WRHALD SISTEM: BHSD EVS ATE SOFTWARE SUBSYSTEM: ASG-151 SOFTWARE SUBSYSTEM TYPE: ATE DESCRIPTION:

Not Available.

ID : 14 BITE : WR-ALD BYSTEM : E-TA AVIONICS ATE BOFTWARE SUBSYSTEM : ANY GSM -185 (B) BOFTWARE SUBSYSTEM TYPE : A E DESCRIFTION:

The E-D Airborne warning and Control System (AWACE) is a high-capacity radar station and command, control, and communication (CD) denter in a modified Bosing 70% airframe. The system includes its associated ground support facilities and additionent. The airborne equipment is composed of the integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer nardware, software, and facilities to support the E-D computer resources. Each version of E-D software consists of over 700 individual computer programs (including programs for Automatic Test Equipment (ATE)).

ATE provides two major capabilities: to determine rapidly whether or not the Unit Under Test (UUT) needs repair and to isolate the fault in a failed UUT to a lower level. ATE software is used for system control, translation, checkout, and execution of test programs to provide ATE Diagnosis/Fault Isolation and Support Software Maintenance.

The AN/GAM-195 is a general purpose, computer-controlled automatic Test System composed of Boeing (B) developed software and Warner Robins (W) developed software. It is used at the depot level in conjunction with adapters, test programs, and orocedures for functional testing and fault isolation of digital and analog/hybrid E-B Avionics circuit cards. In addition, it can be used for on-line generation, editing, and validation of test programs. Basically, the AN/GSM-185 provides power and signal stimuli, measurement, and signal switching to and from the BUT. Through the test program, the computer is used to control and sequence test, to generate but stimulus waveforms, and to analyze but response signals.

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ID : 15 SITE : WR-ALD SYSTEM : E-JA AVIONICS AFE SOFTWARE SUBSYSTEM : ANXISMHIBS(W) SOFTWARE SUBSYSTEM TYPE : ATE DESCRIPTION:

The E-J Airborne warning and Control System (AWACS) is a bight-capacity radar station and command, control, and communication (CD) center in a modified Edeing 707 airframe. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-J computer resources. Each version of E-J software consists of over 700 individual computer programs (including programs for Automatic Test Equipment (ATE)).

ATE provides two major capabilities: to determine rapidly whether or not the Unit Under Test (UUT) needs repair and to isolate the fault in a faired UUT to a lower level. ATE software is used for system control, translation, checkout, and execution of test programs to provide ATE Diagnosis/Fault Isolation and Support Software Maintenance.

The ANVGAM-ISS is a general purpose, computer-controlled Automatic Test System composed of Boeing (B) developed software and warner Robins (W) developed software. It is used at the isolatile in conjunction with adapters, test programs, and independs for functional testing and fault isolation of ligital and analog/hyprid E-D Avionits circuit cards. In eddition, it can be used for on-line generation, editing, and larger of test programs. Basically, the HNVGSM-ISS provides power and signal stimuli, measurement, and signal switching to and from the UUT. Through the test program, the computer is used to control and sequence test, to generate UUT stimulus waveforms, and to analyze out response signals.

THE BDM CORPORATION

15.

ID : 15 SITE : WR-HLC SYSTEM : F-15 SOFTWARE SUBSYSTEM : CC SOFTWARE SUBSYSTEM TYPE : OFP DESCRIPTION:

The F-15, designed and manufactured by McDonnell Aircraft Company (McAir), is a single seat, twin turbofan, air superiority fighter weighing about 40,000 bounds with engines which develop approximately 25,000 pounds of thrust each. It is in the Mach 2.5 class. Armament includes 4 AIM-7 Sparrows, 4 AIM-7 Sidewinders, and a 20mm M-bl gun. The primary mission of the F-15 is air-to-air combat with ground attack as a secondary capability.

The F-15 Central Computer (CC) is an IBM developed general purpose, stored program, simplex, high speed, digital machine designated the AP-1. The CC memory is random access, non-volatile core with a capacity of 16.384~34-61t words (2 parity) which is expandable to 24.576 words.

The F-15 CC Operational Flight Frogram (DFF) is divided into eight program modules which primarily perform mission oriented calculations and output the results to the appropriate F-15 subsystems. The eight program modules are: Executive, Air-to-Air, Air-to-ground, Navigation, Flight Director, Doncrol and Display, Computer Self Test, and Math Subroutine.

ID : 17 SITE : WR-ALC SYSTEM : F-15 SOFTWARE SUBSYSTEM : RADAR SOFTWARE SUBSYSTEM TYPE : OFF DESCRIPTION:

The F-15, designed and manufactured by McDonnell Aircraft Company (McAir), is a single seat, twin turbofan, air superiority fighter weighing about 40,000 pounds with engines which develop approximately 25,000 pounds of thrust each. It is in the Mach 2.5 class. Armament includes 4-AIM-7 Sparrows, 4 ALM-9 Sidewinders, and a 10mm M-51 gun. The primary mission of the F-15 is air-to-air combat with ground attack as a secondary capability.

The radar system consists of Radar Set AN/AFG-63 and Indicator Group OD-60/A. It is a coherent, X-band, multiple FRF, multi-mode, attack radar. The system searches for, acquires, and tracks airborne targets while providing a clutter-tree display of all radar information. The system also provides air-to-ground mapping and ranging, as well as a radar beacon mode in both air-to-air and air-to-ground operation.

The Radar Data Processor (RDP) is a Hughes developed general ourpose computer which provides the focal point for radar set operation as well as for interface with other avionics equipment. The RDF consists of a processor, a special input/output unit and integrated power supply. Three RDF configurations are planned: a low device using core memory, a device using solid state memory, and a larger for solid state device to include the Programmable Signal Processor (RSA Line Replaceable Unit as well as expansion space for the PDR.

The ADF and FSF software are loaded together. The ADF programs provide for radar acquisition, track and built-in-test functions. The FSF programs provide digital processing of the radar returns.

ID : 18 SITE : WR-ALC SYSTEM : F-15 AVIONICS ATE SOFTWARE SUBSYSTEM : ADTS,AIS SOFTWARE SUBSYSTEM TYPE : ATE DESCRIPTION:

Automatic Test Equipment (ATE) software applies collectively to three categories: Test Software. Support Software and Control Software. Test Software includes programs which control the testing operations and procedures (including certification and fault isolation) of the ATE, and programs used to control the stimulus and measurement parameters used in testing the Avionics and ATE Unit Under Test (UUT). Support Software includes programs which aid in preparing, analyzing, and maintaining test software. This software includes HTE compilers, translation/analysis programs, and punch/print programs. Control Software includes programs used during excution of a test program which controls the nontesting operations of the ATE. This software is used to execute a test procedure but does not contain any of the stimuli of measurement parameters used in testing the UUT.

ATE system software primarily includes the Avionics Intermediate Shop (AIS) and Avionic Depot Test Station (ADTS) subsystems. The F-15 AIS ATE support software uses F-15 Adapted ATLAS as the source language, and extensively modified version of the FLACE Compiler, which is called the F-15 Adapted FLACE ATLAS (FAFA) compiler. The control language is Bendix Assembly and the control computer is the Bendix 6200. The executive/OF system controls UUT test operation, self test. CA/Fi tests and mass storage edit functions. The F-15 ADTS Computer Control System (OF system) is divided into five basic functions as follows: (1/ Executive - controls all programs/subprograms. (2) Translation - assembly language to machine language. (3) Execution control - actual test functions. (4) Utility - those routines for editing, debugging, etc. and (5) Maintenance - Self Test and OA/Fi programs.

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THE BDM CORPORATION

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ID : 19 SITE : WR-ALC SYSTEM : JTIDS SOFTWARE SUBSYSTEM : ASIT/OCP SOFTWARE SUBSYSTEM TYPE : D-E DESCRIPTION:

The Joint Tactical Information Distribution System (JTIDS) is an advanced system which provides Communications, Mavidacion, and Identification capabilities in an integrated form for application to military captical and air defense operations. These capabilities are provided through the ability of the system to distribute information quickly and encrypted to provide security and reliability in nostile environments. Security and jamming resistance are obtained through the use of oseudorandom signal processing techniques. The system provides a capability to interconnect scattered sources of surveillance. support, and intelligence information, weapons controllers. weapons systems, and decision-making commanders. UTIUS provides mobile surface and airporne force elements with a relative navigation capability within a common obsidion reference grid and an intrinsic identification capability through the dissemination of trypto-secure position. Velocity. and identity information concerning both friendly and maskile force elements.

The Adaptable Surface Interface Terminal (ASIT) equipment provides a transparent interface between existing ground command and control systems and the JTIDS network. The HSIT includes unique hardware and software along with a SFE regnes Improved Terminal HIT; with an IBM ML-1 Translator Aradessor TER. The HIT will be referred to as the Class I Terminal. The ASIT converts the TADIL B Message Standard of the most platform/system into the Interim JTIDS Message Specification (IJMS) and vice versa.

CD : D: SITE : WR-ALD SYSTEM : UTIOS SOFTWARE SUBSYSTEM : E-TA AWAUS/OUF BOFTWARE SUBSYSTEM TYPE : D-E DESCRIPTION:

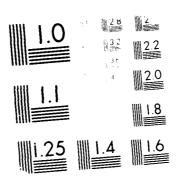
The Joint Tactical Information Distribution System (JTTDS) is an advanced system which provides Communications, Navidation, and Identification dapabilities in an integraced furn for application to military tactical and air defense operations. These capabilities are provided through the ability of one system to distribute information quickly and engrypage is provide security and reliability in hostile environments. Security and jamming resistance are obtained through the use opseudorandom signal processing techniques. The system provides a capability to interconnect scattered sources of surveillance. support, and intelligence information, weapons controllers. weapons systems, and decision-making commanders. UTIDS provides mobile surface and airborne force elements with a relative navigation capability within a common position reference grid and an intrinsic identification capability through the dissemination of drypto-secure position, velocity. and identity information concerning both friendly and hospile force elements.

The Class 1 is a high powered terminal for use in the E-T ind Adaptable Surface Interface Terminal (ASIT) Command. Control and Communications (CT) systems. It consists of pote targular and computer program software to participate in the CT108. The terminal provides the dapability to transmit in assignability sides within the network structure and to receive in all the slots not used for transmission. The Class I Terminal year Interim UTIDS Message Specification (CMS) as its researce tormat.

The EmT and 451T vension of the diass of a recovery of all and of and and an income of the contraction of th

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William Strong and Strong and Strong

ID : 21 SITE : WK-ALC SYSTEM : JTIDS

SOFTWARE SUBSYSTEM : SPYUSER SOFTWARE SUBSYSTEM TYPE : SIM

DESCRIPTION:

The Joint Tactical Information Distribution System (UTIDS) is an advanced system which provides Communications, Navidation. and [dentitication capabilities in an integrated form for application to military tactical and air defense operations. These capabilities are provided through the ability of the system to distribute information quickly and encrypted to provide security and reliability in hostile environments. Security and jamming resistance are obtained through the use of osaudorandom signal processing techniques. The system provides a capability to interconnect scattered sources of surveillance. support, and intelligence information, weapons controllers. weapons systems, and decision-making commanders. JTIDS provides mobile surface and airborne force elements with A relative navigation capability within a common position reference grid and an intrinsic identification capability through the dissemination of crypto-secure position. velocity. and identity information concerning both friendly and hostils force elements.

The Signal Processor User (SP/DSER/ Simulation Software is used to dedug, test and exercise the Adaptable Surface Interface Terminal (ASIT)/F-3 Class 1 Operational Computer Programs (DCPs). It also simulates 3 users (ASIT and/or E-3) or one user and/or a JTIDS network.

ID : DD BITE : WR-ALC BYSTEM : JTIDS

SOFTWARE SUBSYSTEM : SYS EXERCISER

BOFTWARE SUBSYSTEM TYPE : SIM

DESCRIPTION:

The Joint Tactical Information Distribution System (JTIES) is an advanced system which provides Communications. Navidation. and Identification capabilities in an integrated form for application to military tactical and air defense operations. These capabilities are provided through the ability of the system to distribute information quickly and encrypted to provide security and reliability in hostile environments. Security and jamming resistance are obtained through the use of pseudorandom signal processing techniques. The system provides a capability to interconnect scattered sources of surveillance. support, and intelligence information, weapons controllers. weapons systems, and decision-making commanders. UTIDS provides mobile surface and airborne force elements with a relative navigation dapability within a common position reference grid and an intrinsic identification capability through the dissemination of crypto-secure position, velocity. and identity information concerning both friendly and hostile force elements.

The JTIDS System Excerciser (JSE) will be used to fully load the JTIDS net and create test scenarios for correcting problems. The JSE will enable real-time JTIDS tracks to be entered on the net independent from the surface subscriber source.

ID: 23 SITE: WR-ALC SYSTEM: PAVE TACK SOFTWARE SUBSYSTEM: AISF SOFTWARE SUBSYSTEM TYPE: SUP DESCRIPTION:

Fave Tack (AN/AVQ-26) is a 24 hour electro-optical target acquisition, laser/designator, and weapon delivery system for the United States Air Force. The system consists of a fuselage mounted pod and associated cockpit controls and display. It employs an Infrared Detecting Set which permits both day and night operation along with a relative adverse weather capability. In addition, it provides target location data to the aircraft weapons delivery digital computer to permit more accurate delivery of both conventional and guided ordanance. Fave Tack is currently configured for the USAF RF-4C, F-4E, and the F-111F aircraft.

The PAVE TACK Avionics Integration Support Facility (AISF) is an integrated set of hardware and software tools and avionics equipment used for the operational life support of the Pave Tack Operational Flight Program and Operational Test Program as used in the Pave Tack pod. The PTAISF provides capability to analyze impacts of Pave Tack user requests for changes, to modify the Operational Software, to verify and validate the modified software and to generate organizational maintenance level cassettes.

ID: 24
SITE: WR-ALC
SYSTEM: PAVE TACK
SOFTWARE SUBSYSTEM: OFP
SOFTWARE SUBSYSTEM TYPE: OFP
DESCRIPTION:

Pave Tack (AN/AVQ-26) is a 24 hour electro-optical target acquisition, laser/designator, and weapon delivery system for the United States Air Force. The system consists of a fuselage mounted pod and associated cockpit controls and display. It employs an Infrared Detecting Set which permits both day and night operation along with a relative adverse weather capability. In addition, it provides target location data to the aircraft weapons delivery digital computer to permit more accurate delivery of both conventional and guided ordanance. Fave Tack is currently configured for the USAF RF-4C, F-4E, and the F-111F aircraft.

The PAVE TACK Operational Flight Program (OPP) provides an interface function between the PAVE TACK pod, and the associated aircraft cockpit controls/display system and weapons delivery digital computer.

ID : 25 SITE : SM-ALC SYSTEM : F-111D SOFTWARE SUBSYSTEM : WNC SOFTWARE SUBSYSTEM TYPE : DFF DESCRIPTION:

The navigation and weapon delivery system in the fill aircraft is an integrated avionics system. The heart of this system consists of two mission computers. One of these computers functions primarily as a weapons delivery computer and the other functions primarity as a general navigation computer. The Operational Flight Frograms (OFF) loaded into these computers provides the navigation and weapon delivery computations and data required for automatic weapon delivery. Backup logic for most functions of each computer allows either computer to perform both navigation and weapon delivery functions in the event of a single computer failure. Some of the major modules in the OFPs are:

- (1) Navigation
- (2) Data Entry
- (3) Designation
- (4) Steering
- (5) Weapon Delivery
- (6) Air/Air Display

All F-111 DFFs written before 1965 were in IBM Assembly Language and were hosted in IBM 4+Fl computers. Current OFFs are written in Singer CF2-Ex Assembly Language and are hosted in Singer weapons Navigation Computers (WNC). A rewrite of the FB-111A OFF into High Order Language (JOVIAL) is currently in progress.

ID : 25 SITE : SM-442 SYSTEM : F-111F SOFTWARE SUBSYSTEM : WNC SOFTWARE SUBSYSTEM TYPE : OFF DESCRIPTION:

The navigation and weapon delivery system in the F-111 aircraft is an integrated avionics system. The heart of this system consists of two mission computers. One of these computers functions primarily as a weapons delivery computer and the other functions primarity as a general navigation computer. The Or rational Flight Programs (OFP) loaded into these computers provides the navigation and weapon delivery computations and data required for automatic weapon delivery. Backup logic for most functions of each computer allows either computer to perform both navigation and weapon delivery functions in the event of a single computer failure. Some of the major modules in the OFPs are:

- (1) Navigation
- (2) Data Entry
- (3) Designation
- (4) Steering
- (5) Weapon Delivery
- (6) Air/Air Display

All F-111 CFPs written before 1985 were in IBM Assembly Language and were hosted in IBM 4-Pl computers. Current OPPs are written in Singer CP2-EX Assembly Language and are nosted in Singer Weapons Navigation Computers (WNC). He rewrite of the FB-111A OPP into High Order Language (JOVIAL) is currently in progress.

ID : 27 SITE : SM-ALC SYSTEM : FB-111A SOFTWARE SUBSYSTEM : WNC SOFTWARE SUBSYSTEM TYPE : OFF DESCRIPTION:

The navigation and weapon delivery system in the F-111 aircraft is an integrated avionics system. The heart of this system consists of two mission computers. One of these computers functions primarily as a weapons delivery computer and the other functions primarity as a general navigation computer. The Operational Flight Frograms (OFF) loaded into these computers provides the navigation and weapon delivery computations and data required for automatic weapon delivery. Backup logic for most functions of each computer allows either computer to perform both navigation and weapon delivery functions in the event of a single computer failure. Some of the major modules in the OFFs are:

- (1) Navigation
- (2) Data Entry
- (J) Designation
- (4) Steering
- (5) Weapon Delivery
- (b) Air/Air Display

All F-111 OFFs written before 1985 were in IBM Assembly canguage and were hosted in IBM 4-R[computers. Current GFFs are written in Singer CF2-EX Assembly Language and are hosted in Singer Weapons Navigation Computers (WNC). A rewrite of the FB-111A OFF into High Order Language (JCVIAL) is currently in progress.

THE BDM CORPORATION

ID : 18 BITE : CASTLE AFB

SYSTEM : 8-52

SOFTWARE SUBSYSTEM : CPT

SOFTWARE SUBSYSTEM TYPE : ATD

DESCRIPTION:

The B-52 G/H Flight Simulator System Cockpit (caider 48-52 GAT is a uniquely tailored stand-alone software system which operates in real-time, receiving inputs from and transmitting outputs to the Flight Simulator and the Instructor Station.

ID : 29 SITE : CASTLE HFB SYSTEM : B-52 SOFTWARE SUBSYSTEM : WST SOFTWARE SUBSYSTEM TYPE : ATD DESCRIPTION:

The B-SI Weapons System Trainer (WST) will effectively support the training conducted at the Compat Crew Training Schools (SCTS) and Main Operating Bases (MOB) by providing dapapility in initial compat crew qualification, mission qualifications and continuation training for the maintenance of individual crew members and compined drew proficiency in the assigned tactical missions. Training that is directly transferable to the aircraft will be provided for B-SI Filots. Copilots. Navigators. Electronic Warfare Officers, and Defensive Gunners. These capabilities will be implemented through high fidelity simulation of flight and system characteristics and incorporation of instructional features designed to all instructors in their task of instruction and performance evaluation.

Training in mission requirements from preflight to postfillant hav be accomplished and will include:

- (1) Mission planning
- (2) Transition training
- (I) Takeoff and landing training
- (4) Emergency procedure training
- (5) Instrument +light procedures and techniques
- 5) Aerial refueling operations
- General mavigation procedures
- -3) Celestial and pressure pattern navigation
- (9) Gravity weapon delivery
- (10) Special weapon delivery
- (.1) Short Range Attack Missile/Hir Launched Urwise Missile
- delivery
 - (12) Terrain avoidance procedures and techniques
 - 117. Threat evaluation and Hamming
 - (14) Tailgum Firing
 - 15/ Communication procedures
 - (10) Crew coordination
 - 1122 Emergency war order training

ID : 30 SITE : CASTLE AAB SYSTEM : KC-135 SOFTWARE SUBSYSTEM TYPE : ATD DESCRIPTION:

The kC-135 Weapons System Trainer (WST) will effectively support the training conducted at the compat Crew Training Schools and Main Operating Bases by providing capability in initial combat crew qualification, mission qualification, and continuation training for the maintenance of individual crew members and combined crew girectly transferable to the aircraft will be provided for kC-135 pilots, copilots, and navigators. These capabilities will be implemented through high fidelity simulation of flight and system characteristics and incorporation of instructional features designed to aid instructors in their task of instruction and performance evaluation.

The KD-135 WST design is subdivided into four major subsystems: Flight, Navigator, Digital Radar Landmass, and Digital Image Generation (DIG) and includes the following:

- (1) Replica of the kC+1TB flight Station with positions for the pilot and copilot
- (2) Six-degree-of-freedom synergistic motion system for the Flight Station
- (3) Independent replica of the $\pm 0-135$ Navigator Statism with position for the Navigator
 - (4) Modern CRT Instructor Stations
 - (5) Digital Radar Landmass System
 - (6) DIG Visual System
 - (7) Ferkin-Elmer 8/32 computer complexes
- (8) Over-the-shoulder instructor positions on poard each of the two stations

ID : TI BITE : CASTLE HEB SYSTEM : THA TRAINER SOFTWARE SUBSYSTEM : THA SIMULATOR SOFTWARE SUBSYSTEM TYPE : HTD DESCRIPTION:

The T-4 Simulator system simulates the Electronic Warfare (EW cornion of the B-EL. Basically all T4 simulators consist of four major sections in addition to the power distribution panels. The instructor station section, system simulation section, and student station section make up the instructor student equipment. The T4(\forall 5) has four additional student stations and associated power panel.

ID : JD SITE : CO-ALD SYSTEM : F-16 SOFTWARE SUBSYSTEM : FCC SOFTWARE SUBSYSTEM TYPE : CFF DESCRIPTION:

The Fire Control Computer (FCC) performs five functions of primary importance to the F-15 weapon system: weapon delivery. Henergy management, MIL-STD-1550 multiplex bus control. Converse navigation-related functions, and self test. The majority of the logical operations and mathematical computations needed to implement these five functions are carried out in the FCC. Additionally, the FCC interfaces with other subsystems but a the cockpit controls/displays, fuel measurement system, atc., through discrete inputs/outputs and analog inputs/outputs.

The software executed in the FCD is referred to as the FCD Operational Flight Program (CFP). The majority of the FCD OPE is written in the JOVIAL JCB-2 high order language and is mechanized in a modular structure. Minor elements of FCD Loce are written in Majic ToO Assembly language. The FCD OFF is the only F-10 OFF written in high order language. The other six OFFs are written their applicable assembly languages. The FCD OFF is stored in magnetic core memory. The programming languages used were implemented prior to the USAF HOL policy. Cost and timing considerations preclude reprogramming these CFS s in HOL.

The FCC OFF provides logic and computations to implement and integrate fire control system modes and functions. The OFF consists of computer processing instructions which have been developed to satisfy allocated avionic requirements. Secapse of its central role in integrating F-10 sensors and edulament into the desired fire control system, the OFP is designated a configuration item and is managed in accordance with Mile-STD-480 and the configuration management plan LoffiEl.

The FCC OFF is a real-time orburam which councidates decade . . equipment data transfers over the serial didical multiple: data bus and schedules various thodessing activities to include the fire control and havigation modes selected by the pilot.

ID : IT SITE : CO-4LT SYSTEM : F-16 SOFTWARE SUBSYSTEM : HUD SOFTWARE SUBSYSTEM TYPE : CFF DESCRIPTION:

The new Head-Up Display (HUD) system element is a proven electro-optical design configured as a functional follow-on to the HUD currently (1784) installed and operating in the Ambo The HUD is composed of three line replaceable units (LAUS): A display unit, an electronics unit, and a rate sensor Lauch of these are provided by Marconi-Elliott.

The HUD presents to the pilot visual flight and weapon seliver, information as a forward real-world view while operating in a head-up position. The display provides a collimated image upon is optically superimposed on the real-world view.

The display unit includes a combiner plass and bount capable is withstanding an air load of not less than 600 knots, a big: projectness cathode have tube that provides stroke-written symbology refreshed at a 50Hz hate, a hight filter, a stende bandally depressible (0-210 mm) reticle, and selectable a made decrutter control.

The electronics unit processes input data for symbol formula displays. This repertoire includes sympology for 15 astectable operational modes. Hiso, the electronics unit provides the computation of the snapshot air-to-air gunnery solution and the packup missile launch solution.

The rate sensor unit includes the measurement, processing and output of the air trame angular velocity components of roll. Siton, and yaw and normal acceleration as inputs for avionic systems computations.

10 : 74 SITE : 00-ALC SYSTEM : F-16

SOFTWARE SUBSYSTEM : OFT

SOFTWARE SUBSYSTEM TYPE : ATD

DESCRIPTION:

The Operational Flight Trainer (OFT) simulator cocipit is a replica of the F-15 aircraft cockpit. All instruments. *ly-by-wire controls, avionics displays, and indicators (re identical in appearance, color, feel and function to chose of the F-10 aircraft design. All operating controls are monitored by the computational system, and resulting indications will be in response to the real-time software programs and avidnic equipment. A mechanoreceptor during system comprised of a G-Seat, Anti-G Suit, and Seat Shaker is completely integraced into the cockpit to provide the needed motion indications. The fly-by-wire flight controls interact via signal conversion equipment (SCE) with the +light control system software made:. An aural due system will reproduce realistic aircraft sounds. The F-10 GFT is controlled and driven by a NGRD-10/50 computer system. The computer system includes all required SCE. avionics multiplex bus interface, interfaces to visual. tactical, and instructional systems, simulator peripheral equipment, and operational software to control the simulation.

The Fire Control Computer, Stores Management System. Stores Control Ranel, Central Interface Unit, Fire Control Navigation Ranel, Heads-Up Display and Radar Electro-Optics are unmodified aircraft hardware, while the Remota Interface Units and the Fire Control Radar are simulated. Hardware and software produce air-to-air simulation and a representative air-to-ground simulation providing a pseudouround return combined with actual spatial modeling of surface targets. The remaining items of the aircraft avionics equipment. Central Historian Computer. Target Identification Set Laser, Inertial Havigation System and Flight Control computer are realistically sinclated in software to provide the croper stimuling to the control equipment.

ID : IS SITE : 00-ALC SYSTEM : F-16 SOFTWARE SUBSYSTEM : FCR SOFTWARE SUBSYSTEM TYPE : OFP DESCRIPTION:

The F-16 Radar is a coherent pulse-doppler, multimode, digital fire control sensor designed to complement the air superiority and the strike roles of the F+16 multirole fighter. The radar is made up of six line replacable units (LRUs): Antenna, transmitter, low-power radio frequency unit, digital signal processor, radar computer, and radar control pannel.

The Fire Control Radar Operational Flight Program (FCR OFF) resides in the radar computer and interactively controls the functions of the other five LRUs. Communication with the other avionic computers is via the MUX bus. Additionally, the FCR interfaces with other subsystems through discrete inputs/outputs, video, analog input/output, and synchro input/output. The FCR OFF is written in assembly language and the OFP storage medium is erasable-programmable read-only memory.

The FCR OFF provides the processing necessary to implement free radar system modes and functions, and to provide overall control of the radar hardware. The FCR OFF provides the required computations to perform the F-15 modes of operation. The FCR operation mode is determined by mode commands reterved from the Radar Control Fanel, the Fire Control Computer, and the Stores Management Subsystem.

ID : 16 SITE : 50-ALC SYSTEM : F-16 SOFTWARE SUBSYSTEM : 5MS SOFTWARE SUBSYSTEM TYPE : OFF DESCRIPTION:

The Stores Management System (SMS) provides for the munituring, control, release, and jettisching of stores on a selective and emergency basis. The SMS consists of three major components: Stores Control Pannel (SCP), Central Interface Unit (CIU), and the Remote Interface Units. The SCP provides a continuous display of stores identification, location, quantity, percent status, and delivery mode.

Communication with other avionic computers is via one MUX bus. Additionally, the SMS interfaces with other subsystems chrough discrete and analog inputs/outputs. The SMS Operational Flight Program. (DFF) resides in the Clu in grasable-programmable read-only memory and is written in assembly language.

The SMS contains two microprocessors. The first microprocessor provides for the monitor, control and release of the loaded stores, and for jettison on a selective and emergency basis. The second microprocessor updates the SCP display and outputs data to the Bus. If either microprocessor or its associated memory or data busses maifunction, the other microprocessor will shut it down and will take over its functions. The SMS contains the provisions for a comprehensive self-test to minimize the amount of external testing required. The functions provided by the SMS include the following features:

- (i) Monitoring display of store identification, location, quantity and present status.
- (I) Control preparation for stores release through controls which allow pre-programming the SMS on the orders or curing a mission.
- 37 Release accomplishment of armed score release when requirements for release are sacrified.
 - (4) Jettison selective or emergency nebbison or scoles.
- So Mission Loading acceptance of stores invantor, detained SMS memory via the SCF.

A number of pro-determined alternative programs for individual stores are present in the main memory. Each program contains a delection of all of the options (except dactor Arm) necessary to ready a store for release. i.e., a weapon delivery option, an arming option, a release sequence option (such as attendance stores option), and the contains an applicable to all stores of the number of and separation between butter.

ID: 37
SITE: 30-ALC
SYSTEM: F-4
SOFTWARE SUBSYSTEM: MDTS
SOFTWARE SUBSYSTEM TYPE: OFF
DESCRIFTION:

THE BDM CORPORATION

ID : 18 SITE : 00-4LC SYSTEM : F-4E SOFTWARE SUBSYSTEM : ANVARM-101 SOFTWARE SUBSYSTEM TYPE : OFF DESCRIPTION:

THE BDM CORPORATION

ID : TS SITE : 30-MLC SYSTEM : F-46 SOFTWARE SUBSYSTEM : ANVARM-101 SOFTWARE SUBSYSTEM TYPE : GFF CESCRIPTION:

THE BDM CORPORATION

ID : 40 SITE : CO-ALO SYSTEM : F-46 SOFTWARE SUBSYSTEM : LRG-1 +0M SOFTWARE SUBSYSTEM TYPE : LFF DESCRIPTION:

THE BDM. CORPORATION

10: 41 SITE: 00-HUD SYSTEM: MINGTEMAN SOFTWARE SUBSYSTEM: WING II 2015 SOFTWARE SUBSYSTEM TYPE: SIM DESCRIFTION:

Mot Available.

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THE BDM CORPORATION

ID : 42 SITE : DOMALD SYSTEM : MINUTEMAN SOFTWARE SUBSYSTEM : WING VIVESHIS SOFTWARE SUBSYSTEM TYPE : SIM DESCRIPTION:

13: 47
SITE: 00-4L3
SYSTEM: MINUTEMAN
SOFTWARE SUBSYSTEM: WINGS/HS-28
SOFTWARE SUBSYSTEM TYPE: SIM
DESCRIFTION:

10: 44
SITE: 50-ALC
SYSTEM: MINUTEMAN II
SOFTWARE SUBSYSTEM: 35A5/CAFS
SOFTWARE SUBSYSTEM TYPE: SIN
DESCRIPTION:

11 : 45 317E : 30-ALL 337EM : MINUTEMAN 11

SOFTWARE SUBSYSTEM : WING VAHEGARATS

SOFTWARE SUBSISTEM TYPE : SUF DESCRIPTION:

THE BDM CORPORATION

CD: 46 SITE: CO-ALC SYSTEM: MINUTEMAN II SOFTWARE SUBSYSTEM: WING VI HEBURATE SOFTWARE SUBSYSTEM TYPE: SUF CESORIFTION:

THE BDM CORPORATION

ID : 47
SITE : DO-ALC
SYSTEM : RF-4C
SOFTWARE SUBSYSTEM : AN/ARN-101
SOFTWARE SUBSYSTEM TYPE : DFP
DESORIFTION:

10 : 48 BATE : GEHALD BYSTEM : ALOM BOFTWARE SUBSYSTEM : LIT BOFTWARE SUBSYSTEM TYPE : ATE DESCRIPTION:

Level 1 Test (L17) Software performs a field, depot level functional test of the AGM+568 Air Venicle (ALCM). (t supports Missile Shop Replacement Unit Fault leplation, inertial Flatform Calibration and Air Venicle Centification after component replacement or missile upgrade. Major Test Browns are:

- (1) ALCM/Electronic System Test Set Monitoring Circuits verifies that monitoring circuits are within prescribed units prior to applying power to the air venicle.
 - (2) ALCM flower Verifies the air venicle bower tolarance.
- < 3.7 ADCU frogram coad and Verirication coads and tests the load of the Air Vericle Digital unit (ADCU) test software.
- (4) Status and Fault Monitoring Initializes menitoring of the Fault Isolation Record Table for the Level I performance test and verifies proper ADCU hardware operation.
- .5) INE Alignment and Navigation Verifies execution of the INE Ground Alignment Sequence, the performance of the .ME in the Navigation Mode and, that the effects of bias and scale tactor of the vertical accelerometer are within tolerance.
- $_{
 m CO}$. Missile Radar Altimeter (MRH). Common Missile Facar Altimeter (CMRA) Tests the radar altimeters.
- (7) Guided Missile Flight Controller Tests Flight Control of: Flight Control Fower Supply Fegulated (blisde: Movement of Engine thrust Controller: Gain and Frequency desponse of Flight Control Fitch/Foll Channels: movement of Air venicle Clevons and: Aliveness of Telemetry Foints.
- (3) Air Data Element Verifies performance of the Pressure-Bensing Transducer and Electrical resistance Temperature Transmitter.
- .7) Inertial Reference Unit Torquind Test and (NE locerfage Verifies the [NE can torque the stable platfind to tall and pitch and generate the correct phasing of the Filder control roll channel error signal.
- $\times 100$. Finght Control Bensor Test Verities the FCE general electrical power, wheel speed detection circuitry and, finght control enable isolation buffer circuit.
- (1) Main DC Fower. Rotary Switch and Warhead Commands Perities main DC power bus integrity from the DC generator. Separation switch performance and associated circuit integrit is and CHE/Warhead Safe/Arm/Fooling commands.
- -112. Flight Control Arelaunch Test venities alexars are restrained when in a stowed position and that the C.S. Frace \sim Fixet can be successfully conducted.
- 15. Odb Flattorm calibration calibratic sectors overtain there all defended unit error canameters, p.51

ID : 47 BITE : DC-4LU BYSTEM : ALUM SOFTWARE SUBSYSTEM : LFT SOFTWARE SUBSYSTEM TYPE : ATE DESCRIPTION:

Loaded Pylon Test (LPT) Software is a field level functional test to verify operational status of loaded bylon and inartial platform calibration of all missiles, with or without warreads installed.

- (1) Continuity and Isolation verifies that the correct datchodard has been installed, cooling air applied, pylon continuity and isolation and sets the bylon terminal address.
- (2) Fower Application Applies 400 Hz, power to the pylon and to each missile. It also applies electronic power to each missile.
- (7) ADOU Program Load and Veritication + Loads the Hir Vehicle Digital Computer Unit (ADOU) test software and verities that this software has been correctly loaded into the ADOU of each missile.
- (4) INE Flatform Calibration This test is a calibration sequence to update certain inertial reference unit error parameters.
- (5) The Ground Alignment and Navigation verifies capability of the INE to execute properly the INE ground elignment sequence, the performance of the INE in the navigation mode and, that the effects of bias and scale factor of the vertical accelerometer have not exceeded acceptable limits.
- to/ Flight Control Prelaunch Test verifies that the Fld prelaunch test can be successfully conducted.
- -7: Flight Control Tests flight control +unctions including the roll and pitch channel integrators, cruise limit discretes, the DACR and DECR.
- (S) ISU Torquing verifies the Inential Reference entering torquing about the INLE platform roll and bitch acts.
- 7. Tabin Down and Fower Removal is used for giving sets along to memove power from the byton and to vertisy the relative fine fiverface only. MIGS and hissism address.
- (1)) Fast Data Block Screening Test + To detect a cossible MIU modem and bandshake intermittent failures.
- Illy Load Filight Software Reapplies power after the teat program data storage disc has been replaced with the filter software data storage disc and load and initialize the filter software into each missile.
- 12) Fower bown a demotes bower trum the proof of the species said conclusion of the LFT.

ID : 50 SITE : 00-ALC SYSTEM : ALCM SOFTWARE SUBSYSTEM : OFF SOFTWARE SUBSYSTEM TYPE : OFF DESCRIPTION:

The Air Launched Cruise Missile Operational Flight Program (ALCM OFF) provides all of the control logic required for captive alignment, free flight navigation and steering, terrain correlation and warnead control. The OFF executes a unique ALCM mission generated by the strategic mission blanning system and loaded into the Air Venicle Digital Computer Unit (ALCU, by the B-SI Offensive Avionics System (OAS) or B-16 Empedded Computer System. Specifically, the OFF is designed to do the following:

- (1) Provides commands to the Inertial Reference Unit to bring it into coarse alignment.
- (2) Monitors Inertial Navigation Element Built-In-Test circuits and tests avionics subsystems.
- (3) Computes Air Vehicle (A/V) position and velocity relative to earth. $\label{eq:condition}$
- (4) Frocesses independent position measurement data to correct errors in the level channel navigation and inertial element alignment.
- (5) Compares the terrain altitude profile of the H/V flight path to prestored map areas.
- (a) Frowides for safe transition from captive carry to free flight by activating subsystems to provide A/V flight control and propulsion.
- (7). Frovides transitions of mission segments by processing A/V and wavpoint position data.
- (8) Controls mode sequencing of the inertial platform for caging, power down, and nuclear recovery.
- (9) Uses A/V position data and mission data to compute vertical and lateral steering commands.
- (10) Computes the throatile command to maintain the H/L $_{2}$ prestored mach/bime of armival schedule.
- (11) Ferforms warhead arming by processing HVV position and target data to determine arrival at target and activates the warhead at the target via commands.
- $\sim 12)$. Processes temperature and pressure data to compute μ -altitude, mach number, dynamic pressure and weight.
- (13) Restores itself to the pre-event configuration and restarts the system after a nuclear shutdown of the ±0.00 .
- $\pm 14\%$. Commands the H/V to the proper conditions required for deployment of the recover, system for midain retrieval.
- (15). Provides the capability for the OFP to execute in thee flight mode while the 479 is in capilive marry.
- (16) Frowides for the UFF to execute in the avadem -lydgmidaboratory without instructionation data from the carrier archers. F=53

ID: 51 SITE: OC-ALC SYSTEM: B-1B SOFTWARE SUBSYSTEM: CADC SOFTWARE SUBSYSTEM TYPE: OFF DESCRIPTION:

The program requirement of the Central Air Data Computer (CADC) is to supply air data and stall warning data to the primary air data displays, flight control subsystem, avionics computer, and other aircraft subsystems. The basic functions supplied by CADC are as follows:

- (1) Provides altitude and rate of climb/descent calculation.
 - (2) Provides mach/mach rate calculation.
- (3) Provides calibrated airspeed and accoleration calculation.
- (4) Frovides true airspeed/acceleration, and temperature computation.
- (5) Supplies air data and stall warning data to the primary air data displays.
- (6) Provides operational performance data to Centralized Integrated Test System.

ID: 52 SITE: OC-ALC SYSTEM: B-1B SOFTWARE SUBSYSTEM: CITS SOFTWARE SUBSYSTEM TYPE: OFP DESCRIPTION:

The B-IB Centralized Integrated Test System (CITS) provides for the collection and display of fault conditions in both avionics and non-avionics subsystems. Data identifying failures detected by the Central Computing System (CCS), as well as those detected by the TFACU and FACU, are consolidated into a single ACU (GNACU) and transimitted to the CITS dedicated computer. In performing those functions, CITS performs the following tasks:

- (1) Scheduling and timing control of functional modules.
- (2) System error handling.
- * (3) Verification and selection of operator commanded CITS and Avionics Ground Readiness Tests.
 - (4) Transient fault filtering.
 - (5) CITS peripheral and computer self-test.
 - (6) Scheduling and servicing of I/O operations.
 - (7) Reading and recording of operator entered data.
 - (8) Supplying responses to operator request.
- (9) Configuration of aircraft power per load management mode and test requirements.
 - (10) Displays avionics messages.
- (11) Assembles messages for display, print and recording purposes.
- (12) Ferforms fault detection and isolation tests on the aircraft subsystems.

ID: 53 SITE: OC-ALC SYSTEM: B-1B SOFTWARE SUBSYSTEM: EMUX SOFTWARE SUBSYSTEM TYPE: OFP DESCRIPTION:

The Electronic Multiplex System (EMUX) computer program provides for processing necessary to perform electrical control of aircraft subsystems via interface units called remote boxes. The program also provides the interface by which ground maintenance equipment can control and test aircraft subsystems. The EMUX program does not have a functional relationship to other computer programs. The primary functions which are implemented by this computer program are given below.

- (1) Control Box Data Transfer Function This function provides for data transfer within the control box read-write memory. The processing includes transfer of remote box status data and loop-test data.
- (2) CGMI Function This function provides for Central Integrated Test System Ground Maintenance Interface (CGMI) data processing. The function processes three words which contain aircraft subsystem control signals.
- (3) Control Bax BIT Function This function provides for performing the control box self-test functions. The processing consists of testing Boolean processor instructions, the accumulator, the PC Processor and the Caution Data Processor.
- (4) Load Management Function This function provides processing necessary to determine the aircraft electrical load management mode. The function processes signals received from the various aircraft subsystems, the CGMI, and from the EMUX Mode Control function to generate a 4-bit code.
- (5) Caution Light Function This function provides processing in support of the Caution Light Processor (CLP). The function generates a light flashing term value, and three other term values for the equations which the CLP evaluates in its processing.
- (6) Fre/Fost FC Processing Function This function provides for processing in support of the FC Processor. This function is performed before and after FC processing. The purpose of this function is to validate and configure FC data and to initiate the FC Processor. The Post FC processing part of this function provides processing for CGMI indication panel display.
- (7) EMUX Mode Control Function This function provides processing for determining EMUX system mode of operation. There are two EMUX modes: flight mode and ground maintenance mode.
- (8) Aircraft Subsystem Control Functions These functions provide processing for power control of aircraft subsystems. Though the subsystems are different, the processing performed in support of each is virtually the same.

ID: 54
SITE: OC-ALC
SYSTEM: B-18
SOFTWARE SUBSYSTEM: F/CGMS
SOFTWARE SUBSYSTEM TYPE: OFP
DESCRIPTION:

The Fuel/Center of Gravity Management System (F/CGMS) software is divided into two separate programs: Main Frogram and Signal Conditioning Frogram. Together, the Fuel/Center of Gravity Management Frograms provide the processing necessary to measure and display fuel quantity, schedule fuel usage/transfer and control and display aircraft center of gravity. In addition, the F/CGMS processed inputs from the surface position sensors, the Central Air Data Computer (flight regime data), and from the Electronic Multiplex System (EMUX) (discrete inputs from the cockpit controls). It also provides outputs to EMUX (electrical control signals) and to Centralized Integrated Test System (BIT data). The basic functions performed by the F/CGMS software are as follows:

- (1) Frovides an executive function which maintains control of the Operational Flight Frograms (OFPs).
 - (2) Provides an input/output function.
- (3) Provides a fuel quantity correction function which corrects the fuel quantity valve for variations in fuel reference signal, fuel contamination, and fuel density.
- (4) A center of gravity function determines the aircraft Center of Gravity and its limits.
- (5) A fuel usage scheduling function provides for fuel pump and fuel valve controls.
- (6) A test function performs CFU memory, signal conditioners, and F/CGMS system indicators test.
- (7) A ground test function provides ground test routines to perform aircraft ground checkout.
- (8) Development and test software will apply to each module.

ID: 55 SITE: OC-ALC SYSTEM: B-1B SOFTWARE SUBSYSTEM: INS SOFTWARE SUBSYSTEM TYPE: OFF DESCRIPTION:

The Inertial Navigation System (INS) provides the navigational data utilized by the Central Computing System (CCS) and Offensive Radar System. The basic functions of the INS operational software are as follows:

- (1) Real-Time Executive
- (2) Inertial Measurement Unit (IMU) Alignment/Freflight Calibrator
- $\ensuremath{(\mathbb{T})}$ Frovides Navigational calculation information to the CCS
 - (4) Frovides steering outputs to the CCS
 - (5) Frovides IMU Torquing
 - (6) Ferforms Built In Test
 - (7) Frocesses Mux Messages
 - (8) Frovides various utility programs for support
 - (9) Frovides for System Mode Control

ID: 56

SITE : OC-ALC SYSTEM : B-1B

SOFTWARE SUBSYSTEM : ORS

SOFTWARE SUBSYSTEM TYPE : OFF

DESCRIPTION:

The B-1B Offensive Radar Subsystem (ORS) operational flight program is used to control the Programmable Signal Processor (PSP) and the Radar Display Electronics Unit (RDEU) as necessary to accomplish radar functions. The PSP is composed of two functional computing elements, the Radar Computer (RC) and the Array Processor (AP). The RC in general performs the radar control and external interface functions while the AP performs the radar detection data processing. The RDEU contains a general purpose computer which interfaces with the PSP and controls the display of information on the Radar Display Unit. Two basic modes of operation are below:

- (1) Multi-Mode Radar Mode This inclusive mode consists of many modes performing the following functions:
- (a) Frovides quiet mode for non-radiating periods between active modes
- (b) Provides high resolution ground map to provide SAR surface mapping for navigation and target location
- (c) Provides accurate measurement of navigation position errors in Position Update mode
- (d) Frovides low altitude terrain evaluation profiles in Terrain Following mode
- (e) Frovides an azimuth vs. range display of terrain above a specified altitude in Terrain Avoidance mode
- (f) Provides surface mapping for navigation and target location in Real Bean Ground Map mode
- (g) Frovides navigation position update with respect to the known locations of fixed ground beacons in Ground Map Beacon mode
- (h) Frovides azimuth vs. range display of rainfall in front of aircraft in Weather mode
- (i) Frovides interrogation/tracking of airborne beauth for manual rendezvous with bankers in Rendezvous Beaden mode
 - (j) Provides skin-return tracking of other aircraft
- (k) Ground Moving Target [dentification/Tracking mode detects/tracks fast moving ground targe%s in Rendezvous mode
- $\ensuremath{(1)}$ Measures aircraft ground velocity for navigation alignment in Velocity Update mode
- (m) Measures aircraft altitude above terain peyond range of the altimeters in High Altitude Calibrate mode
- (n) Frovides ranging data to a designated ground location in Air to Ground Ranging mode
- (2) Radar Terrain Sensor Mode The following modes, described in the multi-mode radar mode section, are included: quiet, terrain following, terrain avoidance, real beam ground map, position update, and high altitude calibrate.

ID : ST SITE : CO-ALC SYSTEM : BYSC SOFTWARE SUBSYSTEM : BNST SOFTWARE SUBSYSTEM TYPE : AID DESCRIPTION:

THE BDM CORPORATION

10: 56 BITE: JOHALD BYSTEM: BHBD SOFTWARE BUBSYSTEM: FTSS BOFTWARE BUBSYSTEM TYPE: BUF DESCRIFTION:

ID : 59 SITE : DOHALD SYSTEM : BH52 SCETWARE SUBSYSTEM : MCHI EXEC SCETWARE SUBSYSTEM TYPE : DEF DESCRIPTION:

THE BDM CORPORATION

10: 60 SITE: LOHALD SYSTEM: 8-82 SOFTWARE SUBSYSTEM: MOHI EXED SOFTWARE SUBSYSTEM TYPE: DFF DESCRIPTION:

ID : 51 BITE : CC-HLC BYSTEM : E-TA BOFTWARE SUBSYSTEM : INS BOFTWARE SUBSYSTEM TYPE : CFF DEBORIPTION:

The E-J Airborne Warning and Control System (AWACS) is a night-capacity radar station and command, control, and communication (GJ) center in a modified Sceing 7%7 airframe. The system includes its associated ground subport facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground subport equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-J computer resources. Each version of E-J software consists of over 7%0 individual computer programs (including programs for Automatic Test Equipment (ATE)).

The Inertial Navigation System (INS: Computer Program provides navigation information to the Airborne Operational Computer Program through the Control Power Supply. It provides for automatic navigation and guidance through the computations of inertial-sensed accelerations augmented by inputs and guidance from external avionics equipment. It includes the following capabilities:

- $\langle 1 \rangle$. Determine angular orientation and nonizontal velocity of the aircraft.
 - (2) Determine ground speed and drift angle.
 - (D) Determine geographic position.
- (4) Generate and supply a steering signal for control of flight.
- (5) Generate and supply attitude reference signal co flight instruments.
 - (o) Determine the north and east velocity of the sirchaft.

1D : 52 SITE : 55-ALC SYSTEM : E-3A

SOFTWARE SUBSYSTEM : CMEGA SOFTWARE SUBSYSTEM TYPE : OFF

DESCRIPTION:

The E-T Airborne Warning and Control System AWAC3) is a high-tapacity radar station and command, control, and communication (CT) center in a modified Boeing 70% airtrams. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer nardware, software, and facilities to support the E-T computer resources. Each version of E-T software consists of over 700 individual computer programs (including programs for Automatic Test Equipment (ATE)).

The Emega Navigation Computer Program provides corrections to the navigation data in the Inertial Navigation Equipment Computer Program. It provides automatic navigation and guidance through the computations of Omega transmission signals and Doppler velocity measurements augmented by inputs from external avionics equipment. The Emega Navigation Computer program combines the inputs from the Inertial, Emega, and Doppler sensors in a Ralman filter, which then provides computers to the inertial equipment.

ID : ST SITE : CC-ALC SYSTEM : E-DA

SOFTWARE SUBSYSTEM : SMCF SOFTWARE SUBSYSTEM TYPE : OFF

DESCRIPTION:

The E-T Airborne warning and Control System (AWHCS) is a high-capacity radar station and command, control, and communication (CT) genter in a modified Edeing 70% airframe. The system includes its associated ground support facilities and equipment. The airporne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-T computer resources. Each version of E-T software consists of over 700 individual computer programs including programs for Automatic Test Equipment (ATE).

The System Maintenance Computer Program (SMCF) operates in real time with, and under the direct control of, the Aircorne Coerational Computer Program (ACCF) Executive. SMCF execution is interleaved with execution of ACCF tasks and the In-Firent Performance Program.

The SMCP is made up of the following four functions:

- (!) Monitor And Test Subsystem Control Provides fault detection and isolation for the displays of the Identification functional Group and Co-Board Test Monitor and Maintenance Functional Group (OBTM&MFG) through control and interrogation of OBTM&MFG test points. Test sequences and failure priterial are specified by performance monitoring and fault isolation trees.
- (2) Confidence Testing Provides for detection of June 3 Fower Supply (GPS) and avionics interface faults through control and monitoring of what around cests and LFS subsystem status report.
- To Display Dialog Provides for isolation of displational failures by use of a man-machine dialog consisting of displations test patterns, messages, and operator replies via switch actions.
- (4) Central Maintenance Processes equipment scenes reports received from other SMCF functions, the ACCF a equipment of the computer operator. It maintains equipment status tables for use in device allocation and system reconstiturable; initiates display messages and line printer output, holistic coerator of incidents and providing replace health instructions; and reconds maintenance mistory data in magnetic case for data requantion on the dropod.

ID : 64 SITE : 00-ALC SYSTEM : E-JA SOFTWARE SUBSYSTEM : SACA SOFTWARE SUBSYSTEM TYPE : OFF DESCRIPTION:

The E-T Airborne Warning and Control Evstem (AWACS) is a nigh-capacity radar station and command, control, and communication (CT) senter in a modified Boeing 707 air-rame. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-T computer resources. Each version of E-T software consists of over 700 individual computer programs (including programs for Automatic Test Equipment (ATE)).

The Surveillance Radar Computer Frogram (ERCP) operates on the E-D Radar Data Correlator (RDC) and sends preprocessed data to the Airborne Operational Computer Frogram. The RDC consists of a dual processor with separate core program memory and MCS data memory, a special hard-wired processor for pulse doppler range resolution, and an I/O unit for communicating with the radar subsystems and the Control Fower Supply. The SRCP is organized into a main program, normally resident in the RDC, and a fauld isolation test library, which resides off-line on magnetic tape. The SRCP is divided into three functional areas: laca Processing And Control (DFAC), Fault Detection (FD), and Fault Isolation Test (FIT).

- .1/ The DFAC software provides the specific radar functions of input/output (I/O) control and data sequencing, data memory allocation management, mode control, beam stabilization, main beam clutter tracking, range resolution, correlation of radar returns over multiple modulation periods, data processing for pulse doppler, target +ormatting, and a maritime surveillance capability.
- 72) The FD software provides continuous monitoring of various SO/NO-GO fault indications of the radam. Interleaved tests are performed to diagnose faults in the KDC or in the communications links with other radar subsystems. Dedicated time tests and manually selectable tests provide detailed diagnosis of radar units. The FD software controls electron of all tests during turn-on and normal operation. If parameters or test results require it, the FD software controls reconfiguration of the radar by switching in redundant uplications.
- (3) The FIT software consists of detailed tests to solvey radar facilts to replaceable units in major radar solvisten elements. These tests mormalli reside cit-line. and when requested thanvally automatically, are loaded into the Fid.

ID: 65 SITE: GC-4LC SYSTEM: E-JA SOFTWARE SUBSYSTEM: SRGSCF SOFTWARE SUBSYSTEM TYPE: SUP DESCRIPTION:

The E-J Airborne Warning and Control System (AWACS) is a nigh-capacity radar station and command, control, and communication (CJ) center in a modified Boeing 70% airframe. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-J computer resources. Each version of E-J software consists of over 700 individual computer programs (including programs for Automatic Test Equipment (ATE)).

The Surveillance Radar Ground Support Computer Frogram (SRESCE) provides the support software needed to generate, maintain, and test the SRCP and SRMCP. It consists of the following functional components:

- (1) Program Generation Fackage (PGP) Provides for production of the SRCP tapes and maintenance of the radar program files. Includes the RDC assembler and loager.
- (2) Radar Data Generator (RDG) Generates realistic radar target and ECM detection data from a scenario input for exercising the SRCP.
- (3) RDC Functional Simulator (RDCFS) + Simulates the RDC processor and data transfer for active and passive I/O for testing the SRCP on the IBM 370/168.
- (4) Digital Land Mass Blanker Map Tape Generation Program (DLMB MTGP) Generates and/or modifies Land Mass Blanking Maps on magnetic tapes that are read into the DLMB unit.

PGF. RDG. and RDCFS run on the IBM 370/168 and are written mainly in ISM 370 Assembly Language. DLM8 MTGF is written mainly in FORTRAN.

AD : 50 SITE : 00-4LD SYSTEM : 3LOM SOFTWARE SUBSYSTEM : CAS SOFTWARE SUBSYSTEM TYPE : SUB-DESCRIPTION:

BDM/A-85-0510-TR

THE BDM CORPORATION

ID : 97 BITE : SCHHUS SYSTEM : GLOM SOFTWARE SUBSYSTEM : M-DTD SOFTWARE SUBSYSTEM TYPE : SUF DESCRIPTION:

ID : 36 SITE : 30-403 SYSTEM : GLOM SOFTWARE SUBSYSTEM : MET SOFTWARE SUBSYSTEM FYFE : 30F DESCRIFTION:

80M/A-85-0510-TP

THE BDM CORPORATION

10: 57
SITE: DC-ALD
SYSTEM: GLOM
SOFTWARE SUBSYSTEM: DFF
SOFTWARE SUBSYSTEM TYPE: DFF
DESCRIFTION:

BDM/A-85-0510-12

THE BOM CORPORATION

10 : 70 SITE : 00-HLD SYSTEM : GLOM SOFTWARE SUBSYSTEM : WCS SOFTWARE SUBSYSTEM TYPE : DFF DESCRIPTION:

10 : TI SITI : OD-ALO BYSTEM : SHAM BOFTWARE SUBSYSTEM : OFF SOFTWARE SUBSYSTEM TYPE : OFF DESCRIPTION:

IC : TO SITE : TINKER AFF SYSTEM : E-TA SCETWARE SUBSYSTEM : ACCE SCETWARE SUBSYSTEM TYPE : C-E DESCRIPTION:

The E-T Airborne Warning and Control System (AWACS) is a night-capacity radar station and command. Control, and communication (CD) senter in a modified Boeing 70% airtrame. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-T computer resources. Each version of E-T software consists of over 700 individual computer programs including programs for Automatic Test Equipment (ATE)).

The Airporne Operational Computer Program (ACCE) is a compilation of real-time programs chat execute on the airporaft curing an airporne mission. In general, ACCE processes mayigation and radar data, performs calculations, and formats information for display on the E-T consoles. It is written in COVIAL Language and Assembly Linguage and operates on the EAEG 4P, COVI or COVIDATE Processing System. Specifically, ACCE processes data received from all avionics subsystems and from characteridad magnetic tapes to serform, the intended mission. It assists the on-board personnel in the following areas:

- 1 Detect, track, and identify air traffic.
- I: Commit and control weapon resources.
- To Communicate with external interfaces and relay communication messages.
 - 4) Display and process data for on-board operations.
 - E. Ferform on-board training.
- (5) Record data for analysis in the Computer Program Shound Support Center.

ACCE -inctions include the DAFO Executive. Surveillance. Wealons Control. Communications. Displays. Switch Actions. Internal Simulation. Battle Stall, and the System Maintenance Computer Frogram execution. These programs are the very core of the E-D system.

THE SDM COPPORATION

ID : TI SITE : TINKER HEE SYSTEM : E-IA SOFTWARE SUBSYSTEM : UTILITIES SOFTWARE SUBSYSTEM TYPE : SUF DESCRIPTION:

The Z-D Airporne warning and Control System AwACS, is a nightcapacity radar station and command, control, and communication (JD Jenter in a hodified Edeing TD7 airfrage. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the Z-D computer resources. Each version of Z-D software consists of over 700 individual computer programs including programs for Automatic Test Equipment (ATE).

The utility Computer Program LOF provides data for the Airporne Operational Computer Program (ACCF) in support mission deployment. It supports generation, test, and maintenance of ETT computer programs and data bases. The LOF includes the tollowing functions.

Benerates binary map files for display on Situation Display Consoles from alphanumeric inputs consisting of latitude and congulade and display control information. Frovides the capability of presetting values in the HCCF data base. Benerates or ibdates E-T data bases from input adaptation values and JOSTAL COMPOCE symbol definition information for use in the E-T operating environment. Assembles programs written in 4F: Data Endoessing System Assembly Language. Includes seventeen programs and several support routines supporting three functions: control, combilation, and COMFOOL assembly. Benerates program and adaptation tapes for use on the 461 55-. or CC-2 data processor by retrieving program and data elements from secondary storage volumes and transferring them to pagnetit tape in a machine-useable format. Receives as input she magnetic tape containing information produced by the Unit Test Control en Function. Selected portions are formatied and directed to printer for output. Includes a machine instruction level simulator that provides a capability to test 4ft (C-1 or 20-2 computer program units on the CAGSC computer. Includes 4 utility supprogram library of commonly used mathmatical, data conversion, and data manipulation suproutines.

UD : TA SITS : LAMBLE: SYSTEM : UTIDS SOFTWARE SUBSYSTEM : ASITYTECOP SOFTWARE SUBSYSTEM TYPE : UHE DESORIFTION:

The Joint Tactical Information Eistribution System (JT125) is an advanced system which ordyldes Communications. Navidation. and identification capabilities in an integraced form for application to military pactical and air defense operacions. These dapapilities are provided through the apility of the system to distribute information quickly and encrypted to provide security and reliability in hostile environments. Security and jamming resistance are obtained through the use of oseudorandom signal processing techniques. The system provides a dapapility to interconnect scattered soundes of surveillance. support, and intelligence information, weapons controllers. weadons systems, and decision-making commandens. UTIDS chovides mobile auchace and dirounne fonce elements with a relative mavidation capability within a common obsition reterence only and an intrinsip identification capability conduidn the dissemination of crypto-secure position. Relocity. and identity information concerning poth friendly and nostile force elements.

The Translator Processor Idenational Computer Program (1903-software is one part of the Adaptable Eurface Interface Terminal (ASIT system (the other is the ASIT)] ass I Turning and its software, the Communications Processor Doerational Computer Program (CPCS). The TROOP provides a transparent translation of tactical air control message traffic detweet the Interim UTIDS Message Specificacion format and the Cactical in Control System Tactical format, for one to five subscribers. The subcribers are Hir Fince Army ground-based control and faborating Centers or Message Processing Centers.

The TFDER performs several pasic functions in providing the imanaparent interface for the GIT. The primary function of the TFDER is to translate JTDES messages to Europe occupients. The primary functions and to translate 68 messages to JTDES messages. Sther functions include the maintenance of the data pase of the active traces, the implementation of the Joint Tactical winder functions transmission rules, message authoushedgement, and is control of the bransmission and reception of messages. The TRUCE also provides for:

- $1 \leq 5$ Statistics dathert . We have table sets on by use of the foregoing about the ϵ
- ನಿಗಳ ಇಡಲವಾಗವೆ ಅಂತಾ ವರ ಪ್ರಕಾರಿತಿಕೆ ಮುಗಳು ತಿರುವರು ತರ್ಗತ್ ತನ್ನು ಪತ್ರಗಳುಗಳು ತಾಳು ಕಾಳು ಅಂತಾ ಪ್ರಕಾರ ಕಾರುವರು ಅರಣ
- The absilts with solution is not expected to hold seed on the probabilities of the substitution of the solution p_{ij} and the spected solution of the solu

13: 75

SITE : LANGLEY SYSTEM : STRTS

SOFTWARE SUBSYSTEM : STRTS SOFTWARE SUBSYSTEM TYPE : ATD

DESCRIPTION:

The Simulator Tactical Radar Training System (STRTS) is a simulator system to support training requirements of the 407% operational system. STRTS will provide an improved sistem for training and exercising the Tactical Air Control System. Control and Reporting Center/Control and Reporting Fost operations personnel. The STRTS consists of digital computers with peripheral equipment; a video generator; positions for one simulation supervisor technician, six pilot simulators, and a computer operator; automatic inter-system data link equipment; voice communications; and the specifical diagnostic and support software to control system functions. The STRTS can be noused in a van (for portability) or a fixed facility.

ID : To BITE : LANGLEY BYSTEM : TACS BOFTWARE SUBSYSTEM : CAFMS SOFTWARE SUBSYSTEM TYPE : D-E DESCRIPTION:

The Tactical Air Control Center (TACC) is the operations center of the Tactical Air Control System (TACS). The mission of the THOS is to prepare, disseminate, and monitor the execution of coordinated orders for the ampleyment of all the forces assigned, actached, or otherwise mage available to the Air Force Component Commander. The manual TWEC procedures are inadequate to handle the increasing complexity and amount of operational information. Computer Assisted Force Management System (CAFMS) provides an automated assist to the THUS conclor information storage and retrieval along with secure digital communications. A minicomputer located at the TACC will subject up to 12 local terminals and .5 remote terminals. remote terminals have limited stand-alone dapapilities and will be located at the wind Operations Center (WOC). Air Support Operations Center (ASDC), or Control and Reporting Center (CRC). Each terminal consists of a display unit, a veyocar; and a table top brinter. Remote terminals also have a +loccy disc.

The CAFMS software provides automated assistance to two of the TACO functions. Combat Plans and Combat Operations. Combat Plans and Combat Operations. Combat Flans will use CAFMS to construct, review, and disseminate the Air Tasking Order, generate operational mission schedules, and monitor TACO resources. Combat Operations will use IAFMS to follow mission progress, generate recaps and reports, and monitor TACO resources. In addition, CAFMS will expedite information transfer between the TACO and lower equelon elements including the wGC, ASGO, and the CRO.

12 : TT SITE : LANGLE: SYSTEM : TIPI SOFTWARE BUBS:STEM : DD. SR SOFTWARE BUBS:STEM TYPE : D-E DESCRIPTION:

The Tactical Information Processing and Interpretation (4161 System is comprised of several segments. These current, a include the Display and Sontably Storage and Petrieval. Display Engineer: the Imagery Interpretation (41) Segment: and the Manual Radar Reconnaissance Emploitation System (MARAES) Segment.

The DC.SR system is a mobil. sheltered, computer-based Data Management System used by the Intelligence division as their master intelligence data base. The DC/SR software provides the datability for a self-contained (e.g., noncommencial) communications dapability to transmit and receive message e.g., (I) traffic via teletype, digital data link, hard contained voice communication lines. This message flow is essential to the Tactical Air Control Denter (TACC) since it is the only means by which the compat plans division can receive and transmit intelligence information.

The DC/SR segment provides automated assistance to the operational intelligence, collections and targeting functions associated with the TACC pre-planned air tasking order commonation and production cycle. Digital installation and production cycle. Digital installation and croser of pattle data pase files are maintained and updated with the DC/SR. A capacility for weaponeering, and automated report plot generation and dissemination is also available in the segment. Its communication subsystem provides automose assistance for the receipt, processing and transmission of messages, and allows for dedicated gigital data link interfaces with the II segments and the ACTODIN network.

ID : 78 SITE : LANGLEY SYSTEM : TIFI SOFTWARE SUBSYSTEM : II MARRES/TEREC SOFTWARE SUBSYSTEM TYPE : C-E DESCRIPTION:

The Tactical Information Processing and Interpretation (TIFI) System is comprised of several segments. These currently include the Display and Control/Storage and Retrieval (100.68) Segment: the Imagery Interpretation (II) Segment: and the Manual Radar Reconnaissance Exploitation System (MHRRES) Segment.

The II software provides computer assisted interpretation and exploitation of photo imagery reconnaissance data. The mission derived from early 1904 Vietnam requirements for photo interpretation. In 1981 the Tactical Electronic Reconnaissance (TEREC) capability was added to provide Electronic Intelligence (ELINT) at near real-time information processing capability. The MARRES software provides computer assisted interpretation and exploitation of Side Looking Radar Imagery.

The II Segment employs automated light stations and associated ADR equipment and computer programs to expedite the processing and formatting of photographic imagery and ELINT data into meaningful intelligence report. These automated facilities better equip the analyst to fully exploit the increased volume rate and sophistication of present-day imagery by providing the following capabilities: rapid scanning, magnified viewing, and accurate computer aided mensuration of photographic imagery: rapid initial interpretation of photographic imagery to produce intelligence information which is time critial; detailed interpretation and analysis of photographic imagery including collation with data base information collected by other heads; and computer assistance in exploiting the data form MIC-STE-Tel Code blocks.

The dARRES Segment is a basic shelter integrated with equipment, and software designed to be employed by the JSHR for the target carpose of exploiting radar imagenv.

The MARRES has been developed to provide the automated solutionent, sechniques, and procedures required to significance the untailingence dapadrilities of the echelons where in will be employed. The MARRES is designed to accomplish the following basic functions: pre-mission planning and preparation of collated data; analysis, interpretation of ANYUPD-4, tipe redar imageny; demonstich of collated datased in the redar type of the content o

ID : 79 SITE : LANGLEY SYSTEM : 407L SOFTWARE SUBSYSTEM : HUGHES UTIL SOFTWARE SUBSYSTEM TYPE : 5UP DESCRIPTION:

The 407L Computer Programs were designed in a modular fashication facilitate program maintenance and modifications; planned growth items may be added with minimum impact. The state-of-the-art has been advanced in diagnostic programs, tracking logic, on-line site adaptation, and site registration. The net effect is a software package which is easily maintained and capable of modular expansion to meet future contingencies. The 407L programs include operational and recording, utility support for simulation and data reduction, system utility support, and diagnostic troubleshooting.

The Simulation and Data Reduction Programs operate off-line in support of training, system exercise, and program maintenance. The Simulation Programs generate magnetic tape inputs of simulated data which are processed on-line by the Operational Programs. Target reports, switch actions, data link inputs, and tabular data may be simulated. The Data Reduction Programs process data recorded on magnetic tape by the Operational Programs and provide hardcopy output for post-mission depriefing or program checkout.

The Bystem Utility Programs are used for computer program production and maintenance. The Assembler converts symbolic computer instructions into machine code and provides a library of mathematical routines. The remainder of the Utility Programs provide program debudging aids.

The Diagnostic Programs provide on-line System Status monitoring and off-line equipment troubleshooting. The equipment design and design of the diagnostic routines were conducted in concert to provide fault detection capability with isolation to a functional cand droup of from one to ten labor.

ID : 30 SITE : LANGLE' SYSTEM : 407L SOFTWARE SUBSYSTEM : IBM UTIL SOFTWARE SUBSYSTEM TYPE : SUP DESCRIPTION:

The IEM UTIL is essentially a conversion and enhancement of HUGHES UTIL software to an IEM computer environment. It has a functional description similar to the HUGHES UTIL software.

ID : 3: SITE : LANGLEY SYSTEM : 407L SOFTWARE SUBSYSTEM : IORF/IMFF SOFTWARE SUBSYSTEM TYPE : C-E DESCRIPTION:

The 40TE Computer Programs were designed in a modular fashion to facilitate program maintenance and modifications; planned growth items may be added with minimum impact. The state-of-the-rant has been advanced in diagnostic programs, tracking logic, on-line site adaptation, and site redistration. The net effect is a software package which is easily maintained and capable of modular expansion to meet future contingencies. The 407E programs include operational and recording, utility support, and diagnostic troubleshooting.

The Operational and Recording Programs accomplish real-time command and control processing in support of the brime CFC. CFF ACOM mission. The two primary functions controlled by the operational programs are surveillance and weapons control. Burveillance Programs accomplish automatic and materalded-manual tracking, identification processing. crosstell and automatic intersite radar alignment. Flanne's growth features in the surveillance area include automatic skit tracking, and flight plan processing. The Weapons Control Frograms provide automatic guidance computations for offensive and defensive mission (voice) control. The weapons programs accommodate three tactic and profile options structured for tactical air operations. Flanned drowth features in the weapons area are trial intercept computations for weapons assignment, fuel weapons status monitoring, and ground-to-air data link. The General Data Programs accomplish such trigram control and input/output *unctions as executive processing. display generation, switch action processing, data recording. and peripheral equipment input/output processing. The input/putput processing for the automatic data link and ragar processor also falls under general data processing.

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